**Modern Control Theory.** The modern trend in engineering systems is toward greater complexity, due mainly to the requirements of complex tasks and good accuracy. Complex systems may have multiple inputs and multiple outputs and may be time varying. Because of the necessity of meeting increasingly stringent requirements on the performance of control systems, the increase in system complexity, and easy access to large scale computers, modern control theory, which is a new approach to the analysis and design of complex control systems, has been developed since around 1960. This new approach is based on the concept of state. The concept of state by itself is not new, since it has been in existence for a long time in the field of classical dynamics and other fields.

**Modern Control Theory Versus Conventional Control Theory.** Modern control theory is contrasted with conventional control theory in that the former is applicable to multiple-input, multiple-output systems, which may be linear or nonlinear, time invariant or time varying, while the latter is applicable only to linear timeinvariant single-input, single-output systems. Also, modern control theory is essentially time-domain approach and frequency domain approach (in certain cases such as H-infinity control), while conventional control theory is a complex frequency-domain approach. Before we proceed further, we must define state, state variables, state vector, and state space.

Conventional Control Modern Control \* Cinear Sys + nonlihorgs \* Gineur Sys. \* it Can dead with SISO Sys: - Single input Single output SISO, SINO, MISO প্র(১) U(S)\_T and MIN o Sus. Siso SIMO

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2 Tuesday, April 6, 2021 11:13 AM Conventional Conta Modur Control \* S-Domin (root lows Freq Domain \* LTI - Cincor \* toime Domain \* LTI, LTV \* bime in vovent Ses LTV - lineur time > af Ex [m] Vowient \$ 535. Gokg LTJ rocket ZF=mix mix + KX=FE Linear M, K Constants 10700 x it Can deal \* it can't deal with uncertaily uncentente

**State.** The state of a dynamic system is the smallest set of variables (called *state variables*) such that knowledge of these variables at  $t = t_0$ , together with knowledge of the input for  $t \ge t_0$ , completely determines the behavior of the system for any time  $t \ge t_0$ .

Note that the concept of state is by no means limited to physical systems. It is applicable to biological systems, economic systems, social systems, and others.

**State Variables.** The state variables of a dynamic system are the variables making up the smallest set of variables that determine the state of the dynamic system. If at

least *n* variables  $x_1, x_2, ..., x_n$  are needed to completely describe the behavior of a dynamic system (so that once the input is given for  $t \ge t_0$  and the initial state at  $t = t_0$  is specified, the future state of the system is completely determined), then such n variables are a set of state variables.

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**State Vector.** If *n* state variables are needed to completely describe the behavior of a given system, then these *n* state variables can be considered the *n* components of a vector **x**. Such a vector is called a *state vector*. A state vector is thus a vector that determines uniquely the system state  $\mathbf{x}(t)$  for any time  $t \ge t_0$ , once the state at  $t = t_0$  is given and the input u(t) for  $t \ge t_0$  is specified.

**State Space.** The *n*-dimensional space whose coordinate axes consist of the  $x_1$  axis,  $x_2$  axis, ...,  $x_n$  axis, where  $x_1, x_2, ..., x_n$  are state variables, is called a *state space*. Any state can be represented by a point in the state space.

**State-Space Equations.** In state-space analysis we are concerned with three types of variables that are involved in the modeling of dynamic systems: input variables, out put variables, and state variables. As we shall see in Section 2–5, the state-space representation for a given system is not unique, except that the number of state variables is the same for any of the different state-space representations of the same system.

The dynamic system must involve elements that memorize the values of the input for  $t \ge t_1$ . Since integrators in a continuous-time control system serve as memory devices, the outputs of such integrators can be considered as the variables that define the internal state of the dynamic system. Thus the outputs of integrators serve as state variables. The number of state variables to completely define the dynamics of the system is equal to the number of integrators involved in the system.

Tuesday, April 6, 2021 11:13 AM \* State Space Rep. (SSR) = O Non linear, MIMO, time Varient n:- 1/2 & states V = 1/2 & in Pats V = 1/2 & in Pats Non lineu, TU P== # & out Pats X = [X] state Xn] ver states/ -> x,(+) = P. (x,,--, x, u,,-- ur, t)  $\dot{X}_{2} = f_{2}(x_{1}, \dots, x_{n}, U_{1}, \dots, U_{r}, t) U_{r} U_{r}$ 415 (  $X_n = P_n(X_1, \dots, X_n, U_1, \dots, U_r, t)$ (+) = g(x, ..., x, u, , ..., u, t) L'yp yp(t) = gp(x,, ---, xn, U, , ---, Ur, t) Sys  $E_{X'}= (y_{X})+X_{2}+U_{1}+E > f_{1}$ X1 X2  $X_2 = (X_1 + Cb(X_2) + U_{1+}U_{2+}t^3)$  $x_{1}^{3} + x_{2}^{2} + u_{1} + u_{2}^{3}$  $\partial_1 = X_{p+1} \in$  $\partial_2 = X_1 + 3t^2$ Uploaded By: anonymous STUDENTS-HUB.com

5 Tuesday, April 6, 2021  $f(x, u, t) = f_i(x_1, ..., x_n, u_1, ..., u_r, t)$ e e f. f. f. n- E95  $S(x, u, t) = \begin{bmatrix} S_1(x_1, \dots, x_n, u_1, \dots, u_r, t) \\ \vdots \\ S_p(x_1, \dots, x_n, u_1, \dots, u_r) \end{bmatrix}$ PEQS \* Non liner, MEMO, TI (time inversient)  $P(x, u) = \begin{bmatrix} F_{1}(x_{1}, \dots, x_{n}, u_{1}, \dots, u_{r}) \\ \vdots \\ F_{n}(x_{1}, \dots, x_{n}, u_{n}, \dots, u_{r}) \end{bmatrix}$ Sys  $\dot{x}_{1} = (y_{1}) + \chi_{2} + u_{1}$  $X_2 = (X_1 + Cb(X_2) + U_1 + U_2)$ SI = XP J8 6 X1 + X2 4 U1 + 42 1/2 = X, STUDENTS-HUB.com

6 3 MEMO Sys, linear (tim varient) x(t) + B(t) u(t) $\dot{x}(\dot{\mu}) = A(\dot{\mu})$ y(+) = C(+) x(+) + D(+) (1(+) A: - dynamicmatrix, System matrix n: # & states AERNXM BCR<sup>nxr</sup> E# & inputs (input matrix) CER<sup>pxn</sup> Pi=# & outputs (output matrix) DER<sup>pxr</sup> for user matrix B 5 6 10 ) 3 2 (H) A (H) X1 X2 X3 (0 2 1) 3 STUDENTS-HUB.com Uploaded By: anonymous

7 Tuesday, April 6, 2021 11:13 AM \* linew Gys, MIMO, TI I time  $\dot{x}(t) \leq A x(t) + B u(t)$ y(t) = C x(t) + D u(t) $\dot{X} = \begin{bmatrix} z & 3 \\ A \\ L & 5 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$ 2 B U  $\sum_{n \geq 2} \begin{bmatrix} X_{1} \\ X_{2} \end{bmatrix} + \begin{bmatrix} 0 \\ 0 \end{bmatrix}$ 5 MX MO (J MO P. 2(y, , y2

8 Tuesday, April 6, 2021 11:13 AM 6 thotate space Rep.-Ex:if the outputs y = B. y, = Bz  $\mathbf{C}$  $\mathcal{Y}_3 = \mathcal{B}_1$ 130Fs=2  $\mathbf{C}$ n= 2 x Dofs= 2 x 2 = 4, r= 2, P= 3 i Kz X  $x \in R^{4 \times 1}$  $y \in R^{3 \times 1}$  $U \in R^{2 \times 1}$ = A X + BU 4xy 4x J=CX+DU 3x2 3x4 KB. ZM=J, Ö, - B<sup>5</sup>)=2,G  $T_{1} - k_{1}\theta_{1} - k_{2}(\theta_{1} - \theta_{2}) - ($  $\frac{\overline{1}_{1}}{\overline{3}_{1}} - \left(\frac{k_{1}+k_{2}}{\overline{3}_{1}}\right) + \frac{k_{2}}{\overline{3}_{1}} + \frac{c}{\overline{3}_{1}} + \frac{c}{\overline{3}_{1}$  $CB_2 - G_-$ EM= J\_B, K3(  $T_2 - K_3 Q_2 - K_2 (Q_2 - Q_1) - C (Q_2 - Q_1) - C (Q_2 - Q_2) - C (Q_2 - Q$ K2B, 1= 01=> X1 = Brez let(x  $\dot{X}_1 = X_2$ 4×2= B2 => X3 = B2 States 6 . X3 = are 5 Sw esia Stat Eq Da Xz=B, Xy A STUDENTS-HUB.com Uploaded By: anonymous

e, Xz= Bie Xi=Xz Ky 5 Q2 11:30 AN Thursday, Apri  $\frac{k_1+k_2}{\overline{\delta_1}}\theta_1 + \frac{k_2}{\overline{\delta_2}} - \frac{(\dot{\theta}_1 + c\dot{\theta}_2)}{\overline{\delta_1}} - \frac{(\dot{\theta}_2 - c\dot{\theta}_1)}{\overline{\delta_1}} + \frac{(\dot{\theta}_2 - c\dot{\theta}_2)}{\overline{\delta_1}} - \frac{(\dot{\theta}_1 + c\dot{\theta}_2)}{\overline{\delta_1}} - \frac{(\dot{\theta}_2 - c\dot{\theta}_2$ 0,0 (K1+K2) X, Kz X3 - (K3+K2) 1 K2B, - CB2 + Je Je J2 J2 (6) K2+K3 - J2  $X_3 + \frac{K_2}{2} X_1$  $-\frac{\zeta \chi_{4}}{\overline{J_{2}}} + \frac{\zeta \chi_{2}}{\overline{J_{2}}}$  $\overline{\delta_2}$ <u>S</u>2  $\frac{-(k_1+k_2)}{\overline{z_1}} = \frac{c}{\overline{z_1}} \frac{k}{\overline{z_1}}$ T, T2 X c 1 0 24 Kz Õ 482 **4**×4 Jz Xz Xz 5 6 6 0 500 TI ち Ø 3xz 3хч

2.= 8.=  $\frac{\overline{z}}{\overline{z}_1 = \overline{z}_2} - \overline{\omega}$ 10 Thursday, April 8, 2021 11:30 AM XIE -52, X25Q  $\Theta_{z}$ 57.7 5E3= 0  $\chi_3 \leq G_2$ Xy - Br 24562 22- 24- 82 22-2 0 C) B, C, D v) = G1=Z1 1 J2=B 72 20 yz = B = = 23  $\mathcal{O}$ STUDENTS-HUB.com Uploaded By: anonymous

11 Thursday, April 8, 2021 11:30 AM 8+K mit+Ci+KX=5 nan Dynami, Cal Sys MEMO, SI 10, MT. 50, SSS , K.J. . . 55 X 25 X Cinn EF=mx X2-652 -cx-P=mx KX2 P= - (m "x + C x 1 EM = J.B 7 - Pri - Kx2 (2 = 50 B T-(mx+cx)r-kBr2 = J08 Tx - (mB(1 + C'BK)) ~ - KB12 = J5B 7 STUDENTS-HUB.com

12 11:30 AM Krz April 8, 2021 S 5+1 GIDiefeq. Y=X n=2V= 1 P= | let tu states ave \_ Z1 = B => Z1 = B Z = BZ Z2  $\frac{1}{72} = -(x_1^2 + z_2 - k_1^2 + z_1)$   $\frac{1}{72} = -(x_1^2 + z_2 - k_1^2 + z_1)$   $\frac{1}{72} = -(x_1^2 + z_1 - k_1^2 + z_1)$ ii II +T (mr<sup>2</sup> リニ Targu y=c 20 0-482 24 (m r,2+5) A  $(mr_{2}^{2}4)$ X=Qr D = Ϋ́, Ο 2+ Y---STUDENTS-HUB.com

13 <u>Eigenvalues and Eigenvectors</u> Tuesday, April 13, 2021 11:26 AM D'An eigenvalue & A mat/ix is a nonzero vector q in R Such that Aq = 29, for Some Scelart. ② An eigenvalue of A is a Scalar > Such that Use equalion Aq = >q has a non-trivial Solution x; fAq = 19 for q we say that I is the eigenvalue for q and that q is an eigenvector for 2 Notes: S-Eigenvectors are by defindion are non Zero vectors - Eigenvalues many be equal to Zero Aq = 2 q Exiz X = [] X + [] JU find eigen values and eigen vectors? Also study the Stability Alless =>  $Aq^2 = \lambda q^2 = \lambda q^2 - \lambda q^2 = 0$  $(A - \lambda^{I})\overline{q} = 0 \quad (A - \lambda^{I})\overline{q} = 0$  $(A - \lambda I) = \overline{0} = (\lambda I - A) = 0$ STUDENTS-HUB.com

14 Tuesday, April 13, 2021  $|(\lambda I - A)| = 0$ A= 50 1 1 2 E  $\begin{array}{c} \cdot & \circ \\ \cdot & \circ \\$ X-7  $\lambda^2 - 2\lambda - 1 = 0$ :0 => Lz= 2.4142  $\lambda_1 = -0.4142$ S-domain Unstable stable Unstable w.H w.4 Jublicution nliculion 0 m. stable Sizetd 51,2= 10 53,4 = +1 unstable mistable STUDENTS-HUB.com

15 Tuesday, April 13, 2021 11:2  $A : \begin{bmatrix} 0 & 1 \\ 2 & 2 \end{bmatrix} \Rightarrow \begin{bmatrix} 1 & -A \end{bmatrix} = \begin{bmatrix} 1 & -1 \end{bmatrix} = \begin{bmatrix} 1 &$  $q_1 = \left( \begin{array}{c} q_{11} \\ q_{12} \end{array} \right)$ [2] [A-I]], =0  $\begin{bmatrix} -0.4142 & -1 \\ -1 & -2.4142 \end{bmatrix} \begin{bmatrix} 9_{11} \\ 9_{12} \\ 9_{12} \end{bmatrix} = \begin{bmatrix} 0 \\ 6 \\ 0 \\ 0 \end{bmatrix}$ the rank = 2 So we have two Eqs with two variables if the Nomk 51 so we have one equete det (I.I-A) \$ \$ 0 Pull rowk (= 0 not full rowk det(4]-A)= -0.4142x-2.4142 - 1= 0 So rank=1 1 pt take 1st row -o.4142 911 - 912 = 0  $et 9_{n=1} = -0.4142(1) = 9_{12}$  $\overline{q}$ :  $\left[-0.4142\right] \Rightarrow \overline{q}$ :  $\left[-0.4142\right]$ no (mulization of vector normalized vector  $\left(-0.4142\right)^2$ Uploaded By: anonymous STUDENTS-HUB.com

16 Tuesday, April 13, 2021 11:26 AM for 2 = 2.4/42  $\left[ \lambda_{2}I - A \right] \widehat{q}_{2}^{2} = \widehat{\phantom{q}}_{2} = \sum \left[ \lambda_{1} I - A \right] = \left[ \begin{array}{c} \lambda & -1 \\ -1 & \lambda_{-2} \end{array} \right]$ 2.4142 -1 -1 0.4142  $\left[ 1_2 \left[ -A \right] = \right]$ det (12I-A)= -2.4142×0.4142-1=0 1et leake Second row -92140.4142922=0 let 922=) 921 = 0.4142  $\overline{9}_2 = 0.4142$ etz =  $\frac{1}{(0.4142)^{2} + (1)^{2}}$ in mat lab rank (A) "to check the rank of matrix - Jet(A) " to check the detriminant & A. Lamder - eigendalus [amder] = eig A " eigenvectors + eigenvalug" Uploaded By: anonymous STUDENTS-HUB.com

17
Tuesday, April 13, 2021 11:26 AM

18
Tuesday, April 13, 2021 11:26 AM