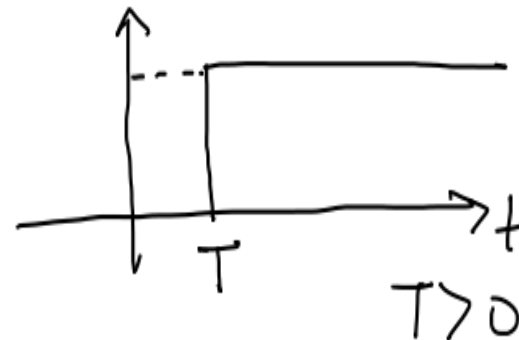
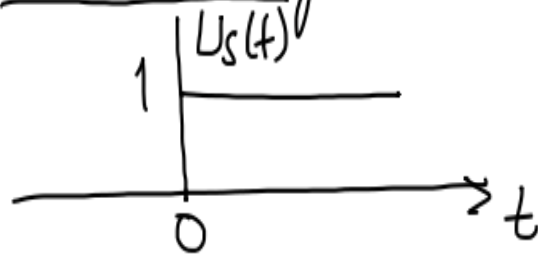


Some properties of Laplace Transform

1. Linearity: Laplace Transform is a linear transform:

$$\mathcal{L}\{a_1 f_1(t) + a_2 f_2(t)\} = a_1 \mathcal{L}\{f_1\} + a_2 \mathcal{L}\{f_2\}$$

2. Time-delay:



$$\mathcal{L}\{f(t-T) u_s(t-T)\} = e^{-sT} F(s)$$

3. Differentiation:

$$\mathcal{L}\{f'(t)\} = sF(s) - f(0)$$

4. Integration:

$$\mathcal{L}\left\{\int_0^t f(\tau) d\tau\right\} = \frac{1}{s} F(s)$$

5. Final value Theorem:

$$\lim_{t \rightarrow \infty} f(t) = \lim_{s \rightarrow 0} sF(s)$$

only valid if all poles of $sF(s)$ are in the LHP.

Ex: $F(s) = \frac{5}{s(s^2+s+2)}$

$$f(\infty) = ?$$

$sF(s)$ must be a stable TF?

$$sF(s) = \cancel{s} \frac{5}{\cancel{s}(s^2+s+2)} = \frac{5}{s^2+s+2} \leftarrow \text{stable}$$

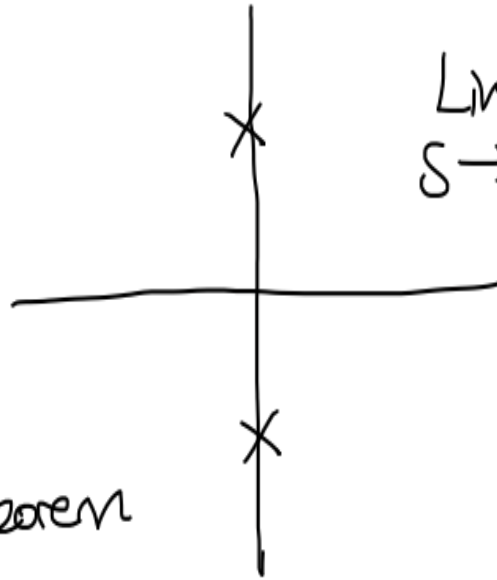
$$\Rightarrow f(\infty) = \lim_{s \rightarrow 0} sF(s) = \frac{5}{2}$$



EX: $F(s) = \frac{4}{s^2+4}$, $f(\infty) = ?$

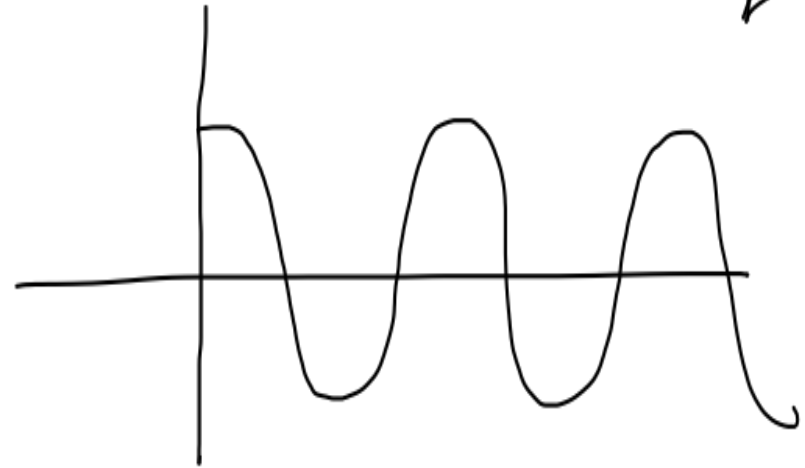
$$sF(s) = \frac{4s}{s^2+4}$$

we can not apply
final value theorem



$$\lim_{s \rightarrow 0} sF(s) = \frac{4s}{s^2+4} = 0$$

do not
match



6. Initial Value Theorem:

$$\lim_{t \rightarrow 0^+} f(t) = \lim_{s \rightarrow \infty} s F(s)$$

no stability cond'n
is required.

EX: $F(s) = \frac{5}{s(s^2 + s + 2)} \Rightarrow \lim_{t \rightarrow 0^+} f(t) = \lim_{s \rightarrow \infty} s F(s) = 0$

EX: $F(s) = \frac{4}{s^2 + 4} \Rightarrow \lim_{t \rightarrow 0^+} f(t) = \lim_{s \rightarrow \infty} s F(s) = 0$

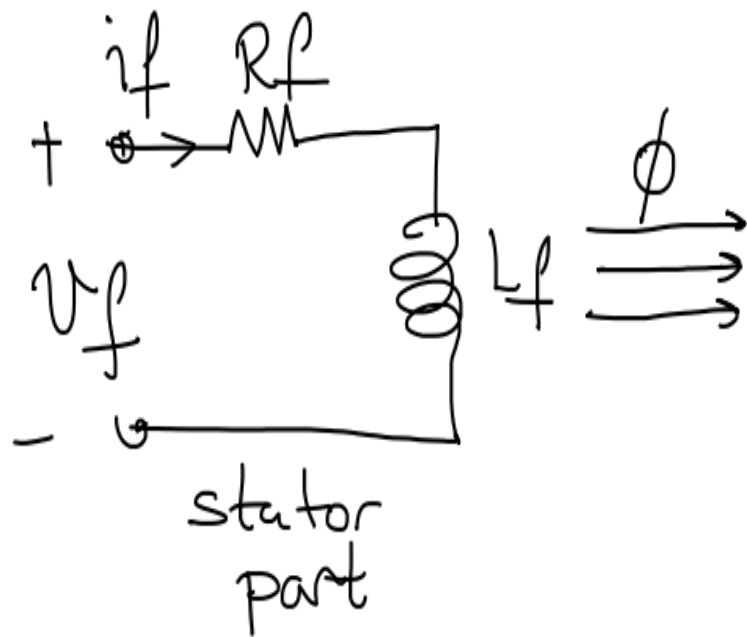
7. Convolution: $F_1(s) = \mathcal{L}\{f_1(t)\}$, $F_2(s) = \mathcal{L}\{f_2(t)\}$

$$\mathcal{L}\{f_1(t) * f_2(t)\} = \mathcal{L}\left\{\int_{-\infty}^{\infty} f_1(t-\tau) f_2(\tau) d\tau\right\}$$

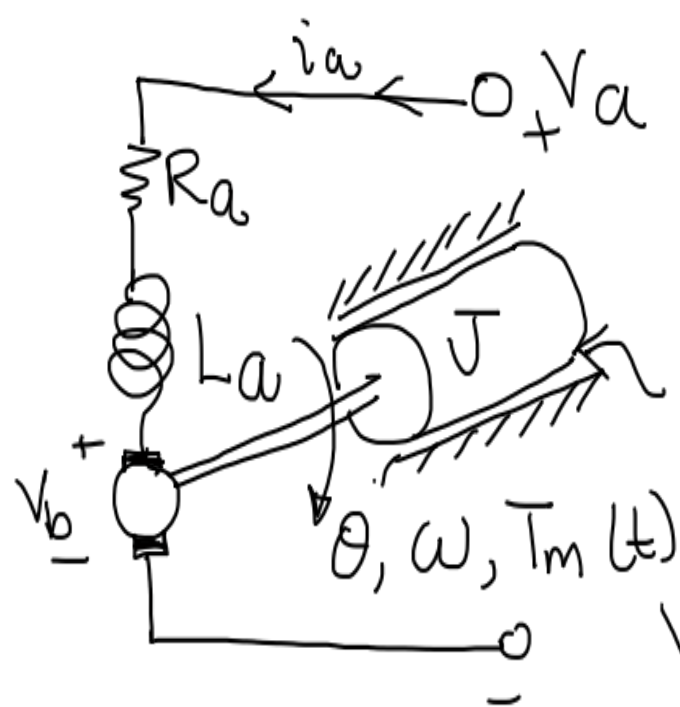
$$= \mathcal{L}\left\{\int_{-\infty}^{+\infty} f_1(\tau) f_2(t-\tau) d\tau\right\} = F_1(s) \cdot F_2(s)$$

Ex: (Transfer function of DC Motor)

- A DC motor converts direct current (DC) electrical energy into rotational mechanical energy.
- Because of its features, such as high torque, speed controllability in a wide range, portability, well-behaved speed-torque characteristics, DC motors are widely used in numerous control applications such as robotic manipulators, transport systems, disk drives, machine tools etc.



Field part can replace this part by a permanent magnet.
 \Rightarrow PMDC motor



L_a : armature inductance of coil
 R_a : armature resistance

$$V_b = K_b \cdot \omega$$

i_a : armature current

Recall: A coil placed in a magnetic field undergoes a torque

$$\tau = K_i \cdot \phi \cdot i_a$$

moment (torque) constant

If i_f is constant, we have an armature controlled DC motor. The torque $T_m(t)$ is given by

$$T_m(t) = \underbrace{K_i K_f i_f}_{\text{const.}} \cdot i_a = \underbrace{K_m}_{\substack{\text{moment (torque)} \\ \text{const.}}} \cdot i_a(t)$$

$$\boxed{\begin{aligned} L \frac{d}{dt} i(t) &= V_L(t) \\ L s I &= V \end{aligned}}$$

$$\textcircled{1} T_m(t) = K_m i_a(t)$$

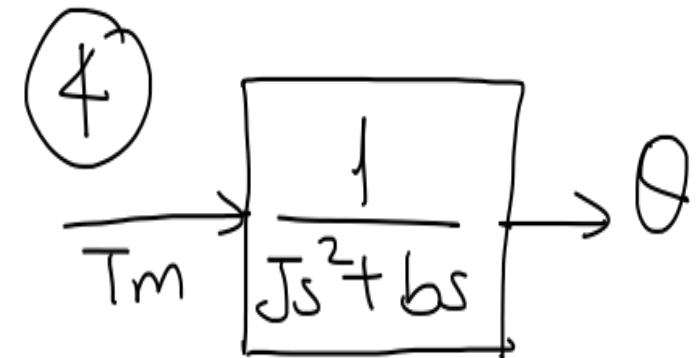
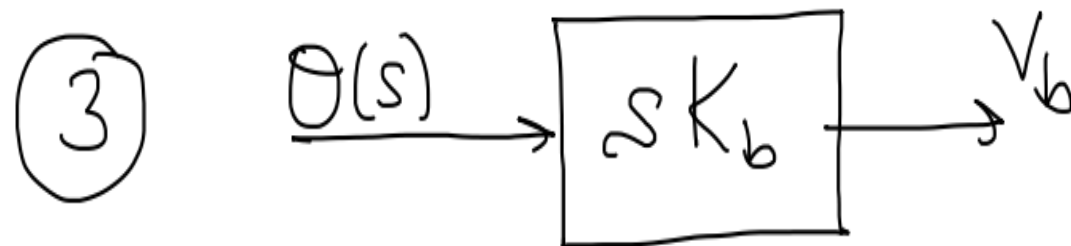
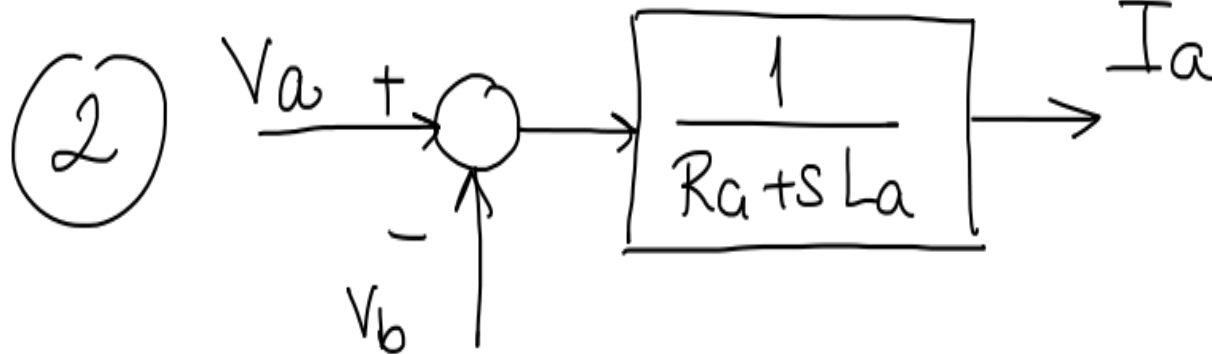
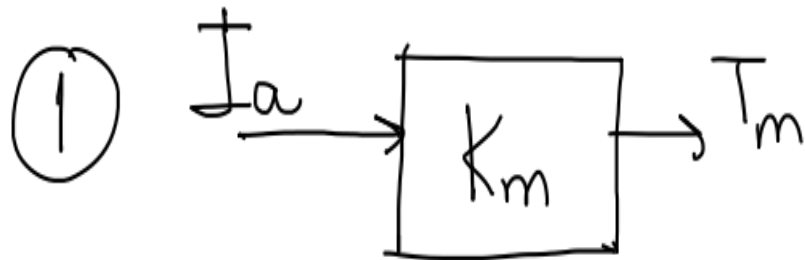
$$\textcircled{2} V_a(s) = (R_a + s L_a) I_a(s) + V_b(s)$$

Back EMF Voltage
generated by the
rotation of the
shaft $\approx \omega$

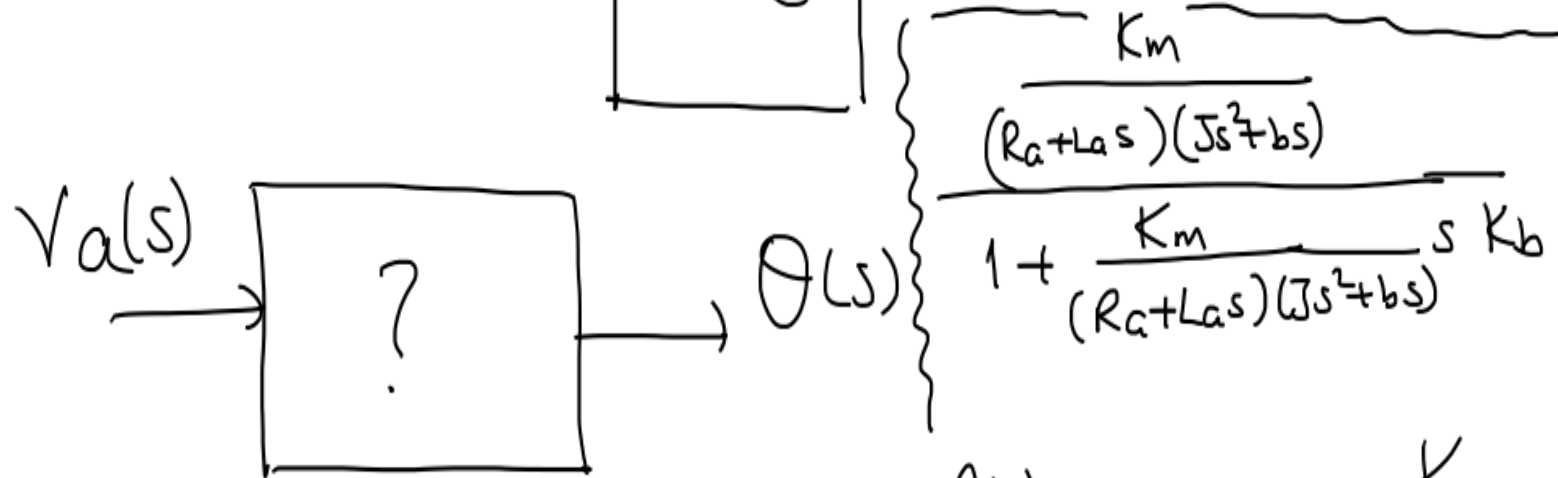
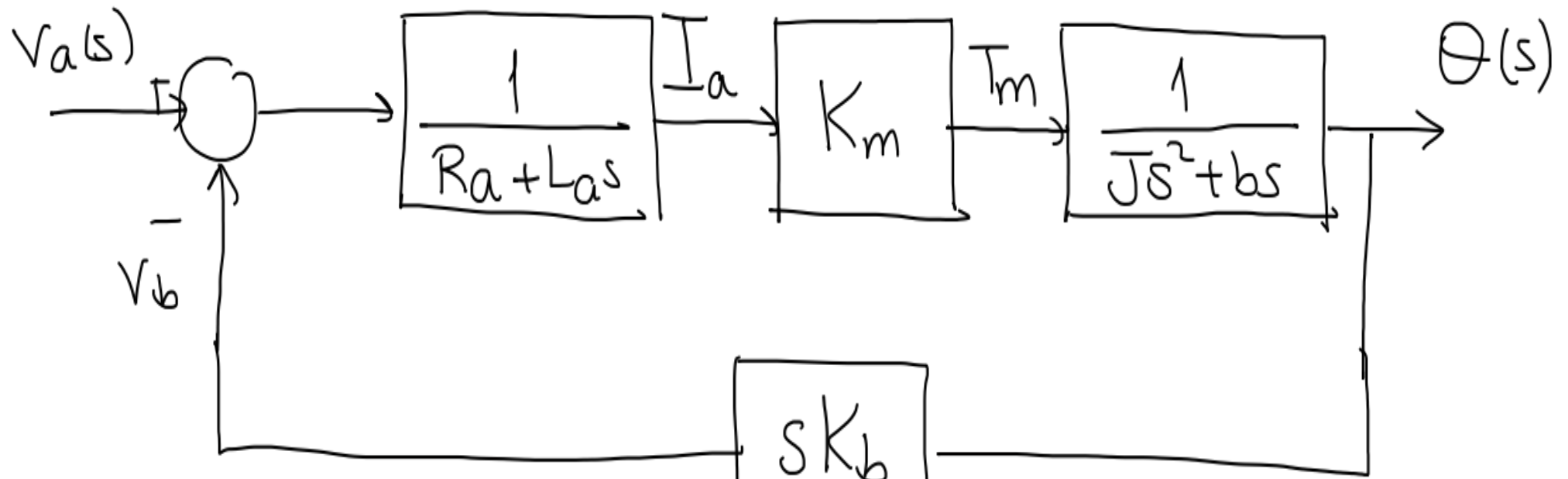
$$\textcircled{3} V_b(s) = K_b \omega(s) = K_b s \theta(s)$$

④ $T_m(s) = (Js^2 + bs) \Theta(s)$ b : viscous friction const.

J : inertia of the motor

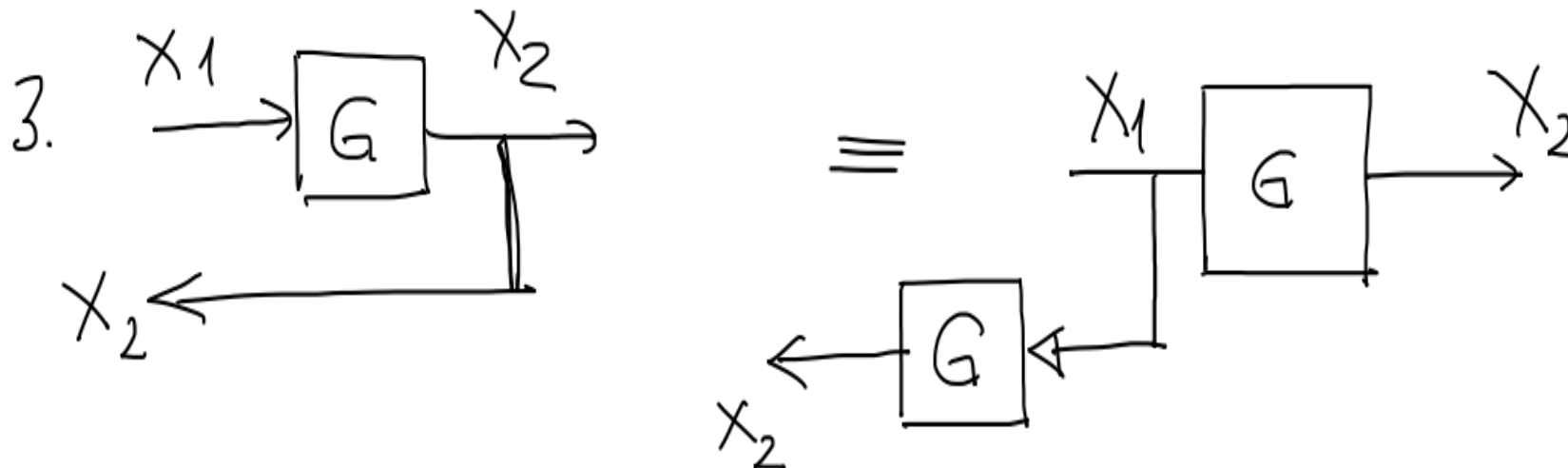
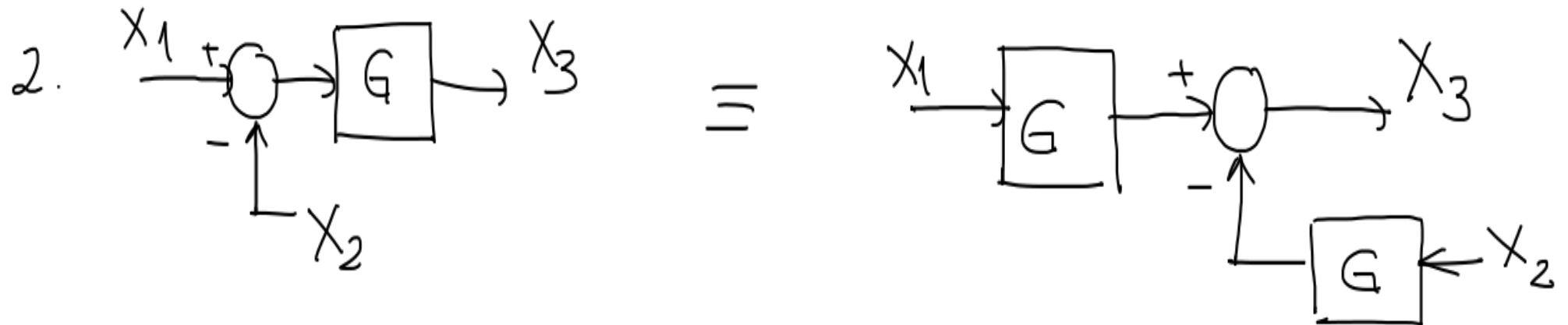
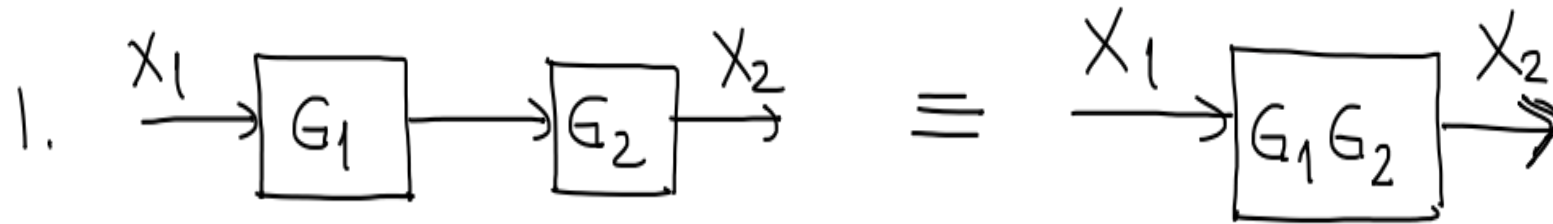


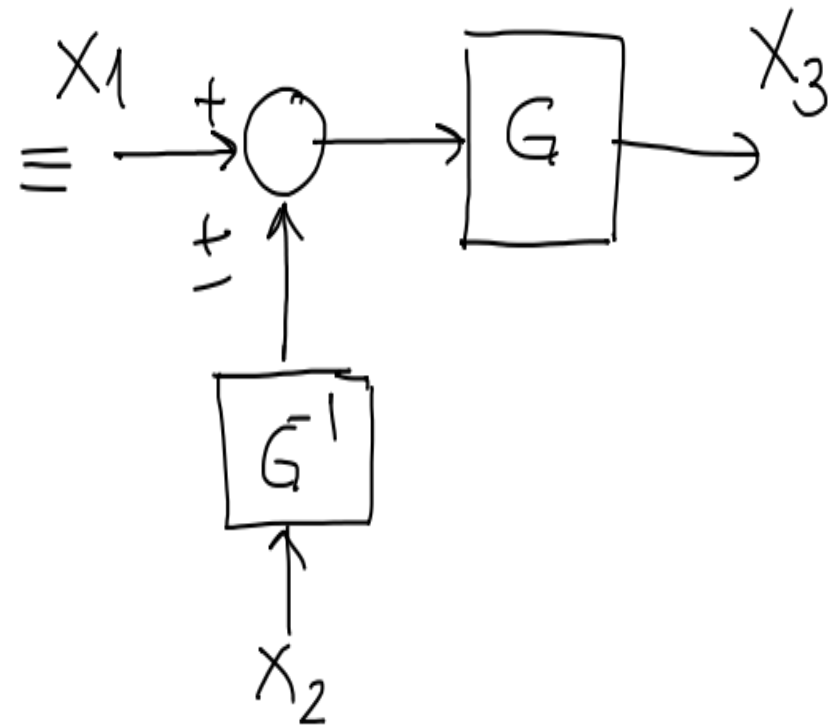
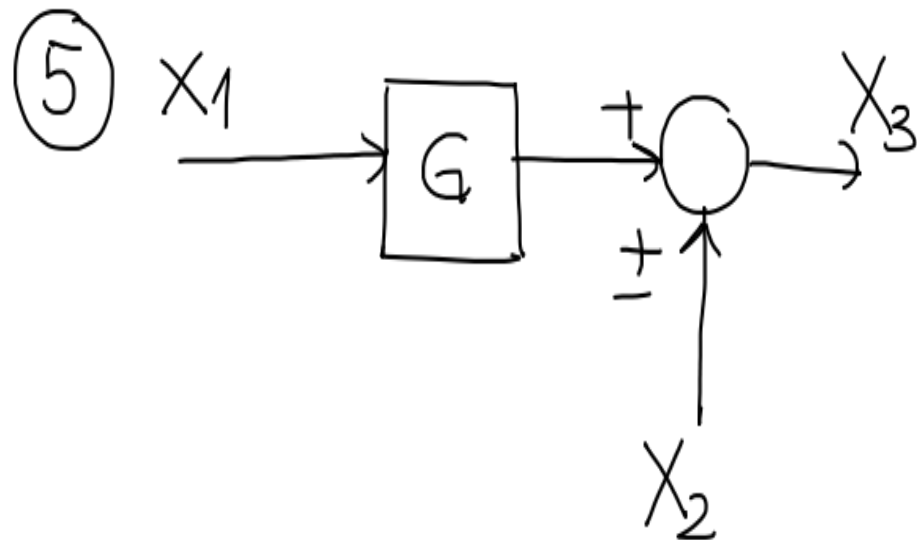
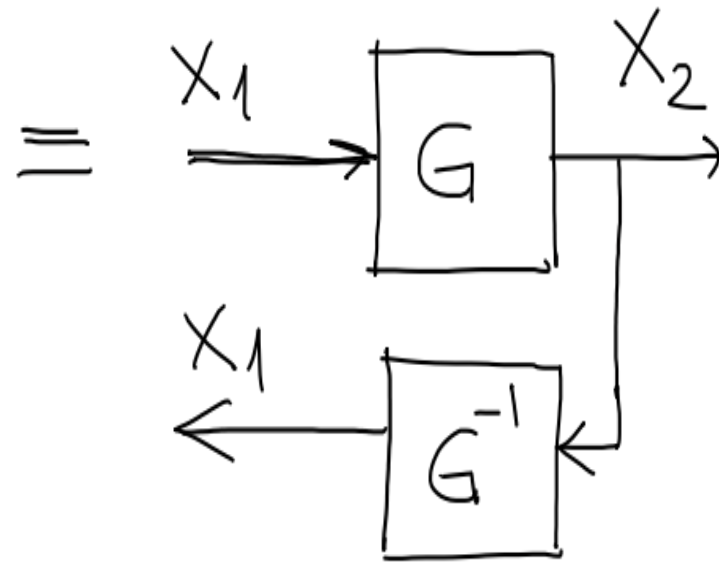
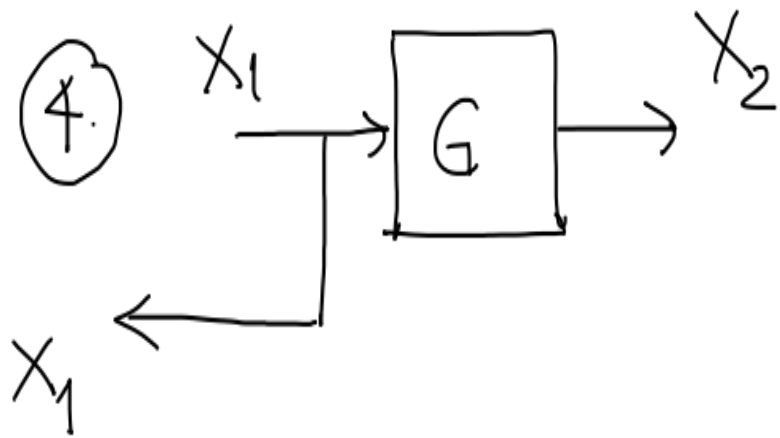
Block diagram of the system:

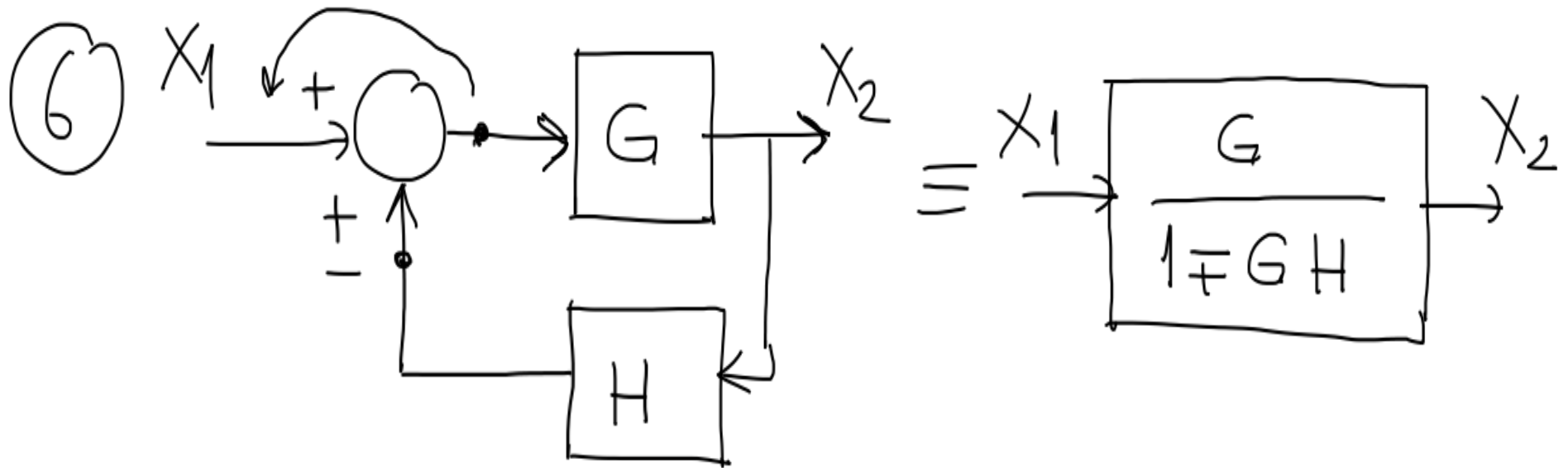


$$G_m(s) = \frac{\Theta(s)}{V_a(s)} = \frac{K_m}{s[(R_a + L_a s)(J s + b) + K_b K_m]}$$

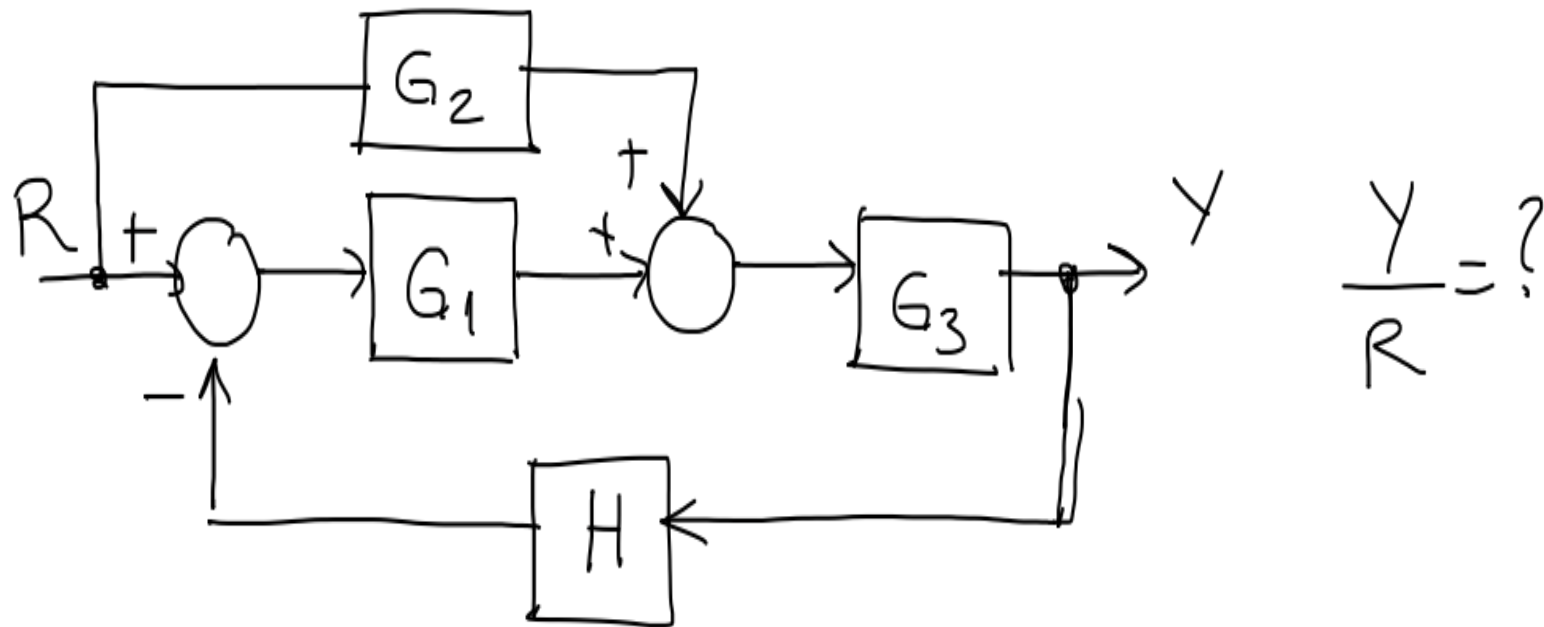
Block Diagram Reduction







$$X_2 = G(1 - GH)^{-1} X_1$$

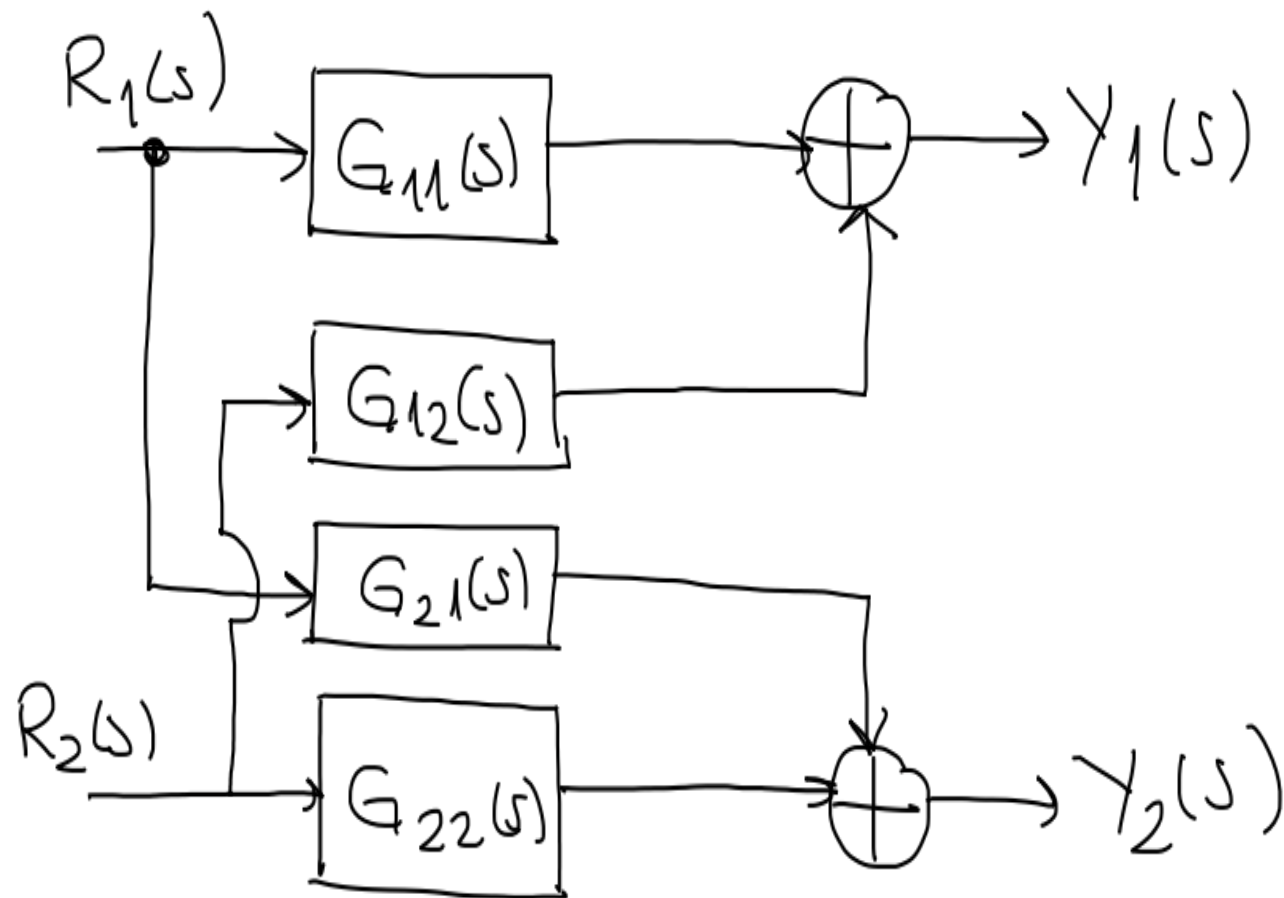


$$Y = G_3 G_1 (1 + G_1 G_3 H)^{-1} R + G_3 G_2 R$$

$$\frac{Y}{R} = \frac{G_3 G_1}{1 + G_1 G_3 H} + G_3 G_2$$

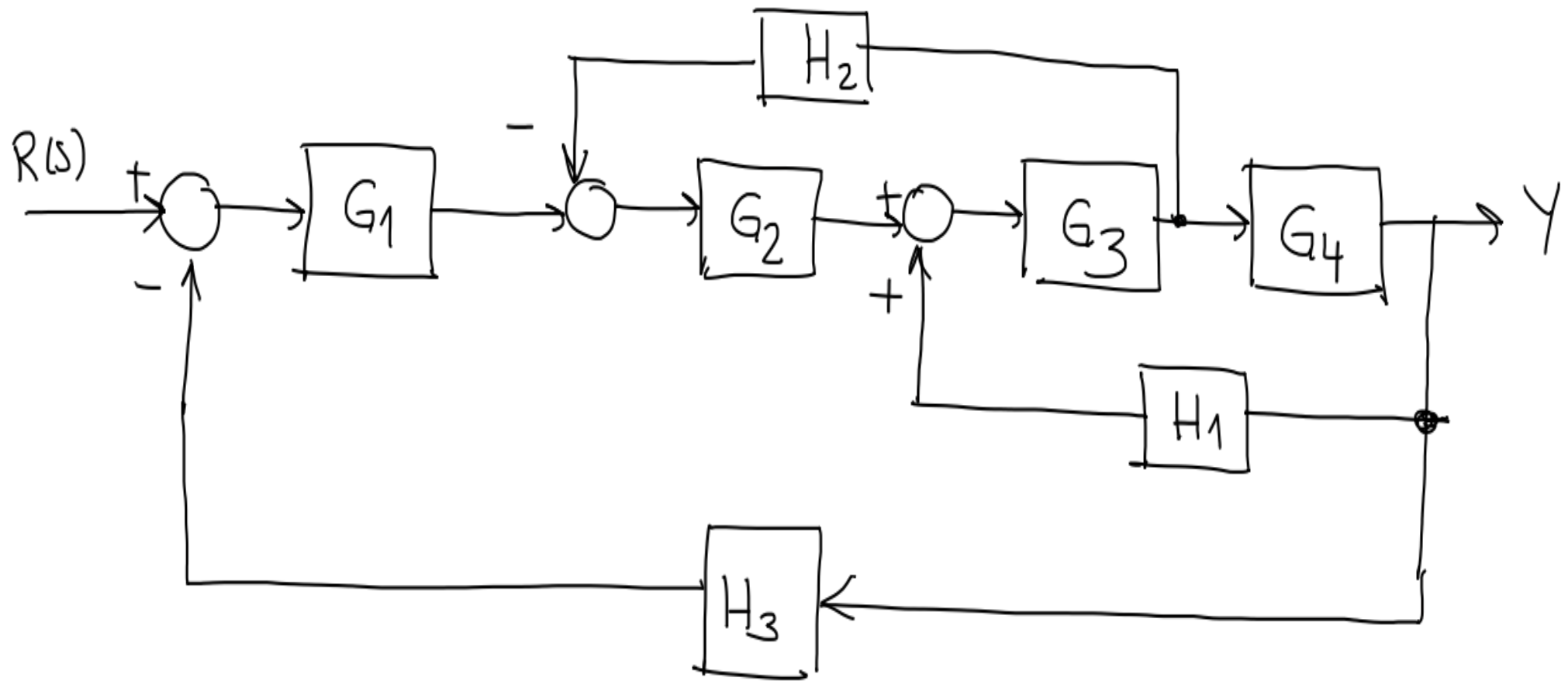
If our system has MI and MO?

EX:

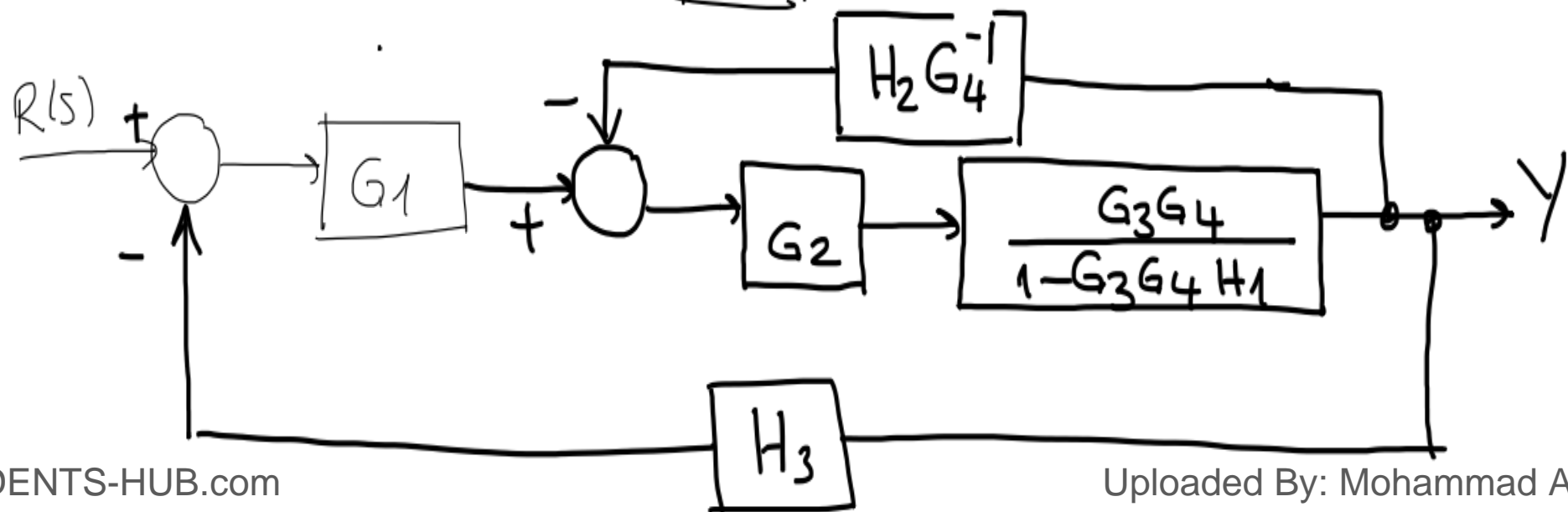
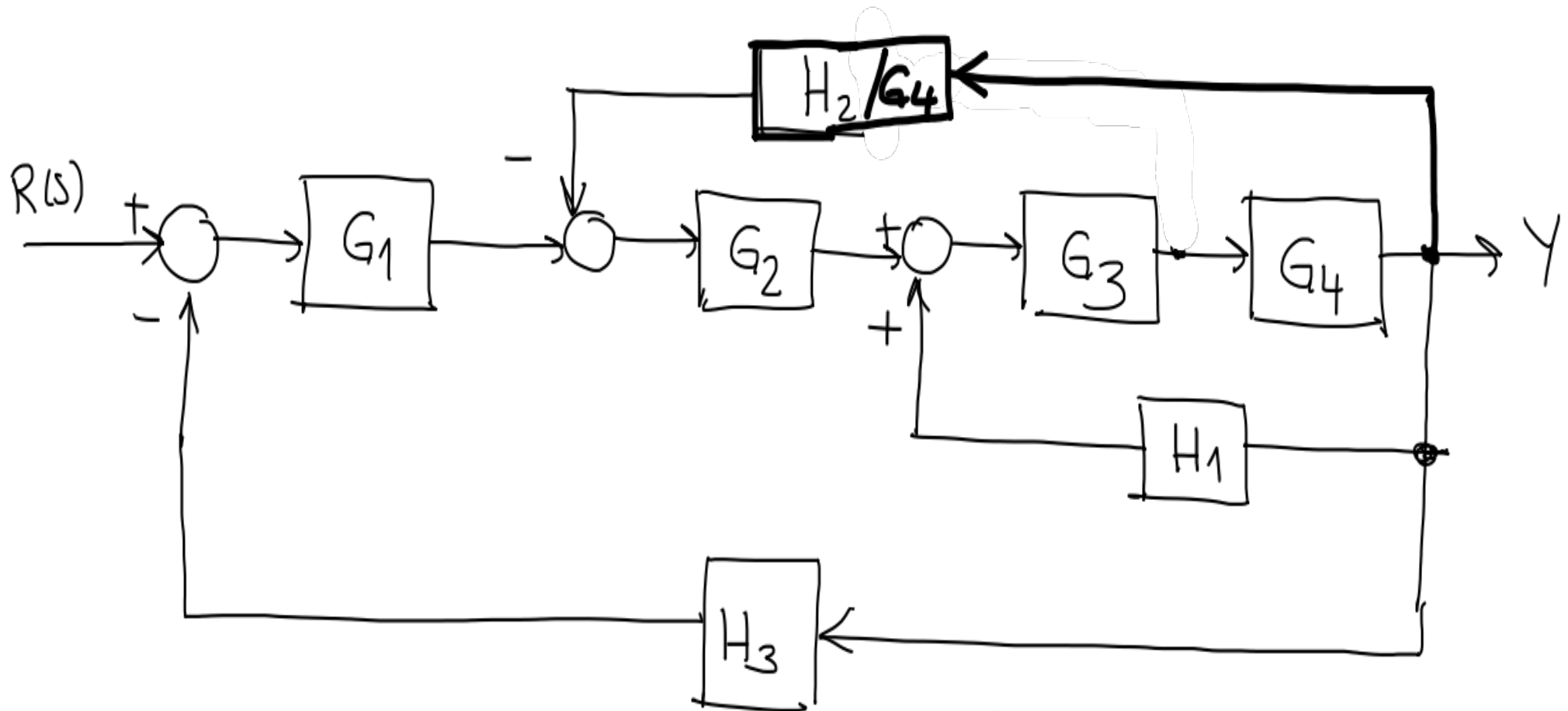


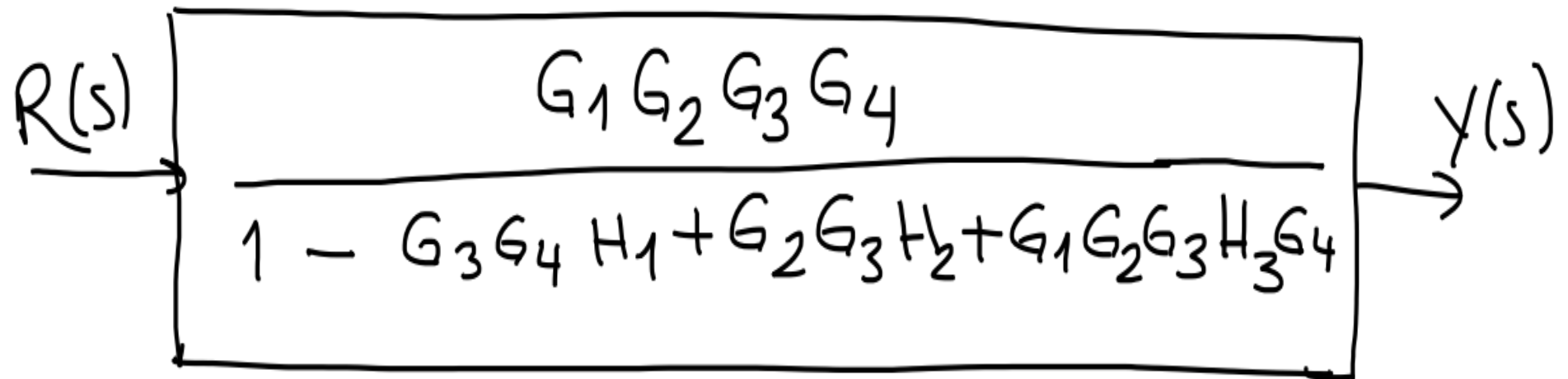
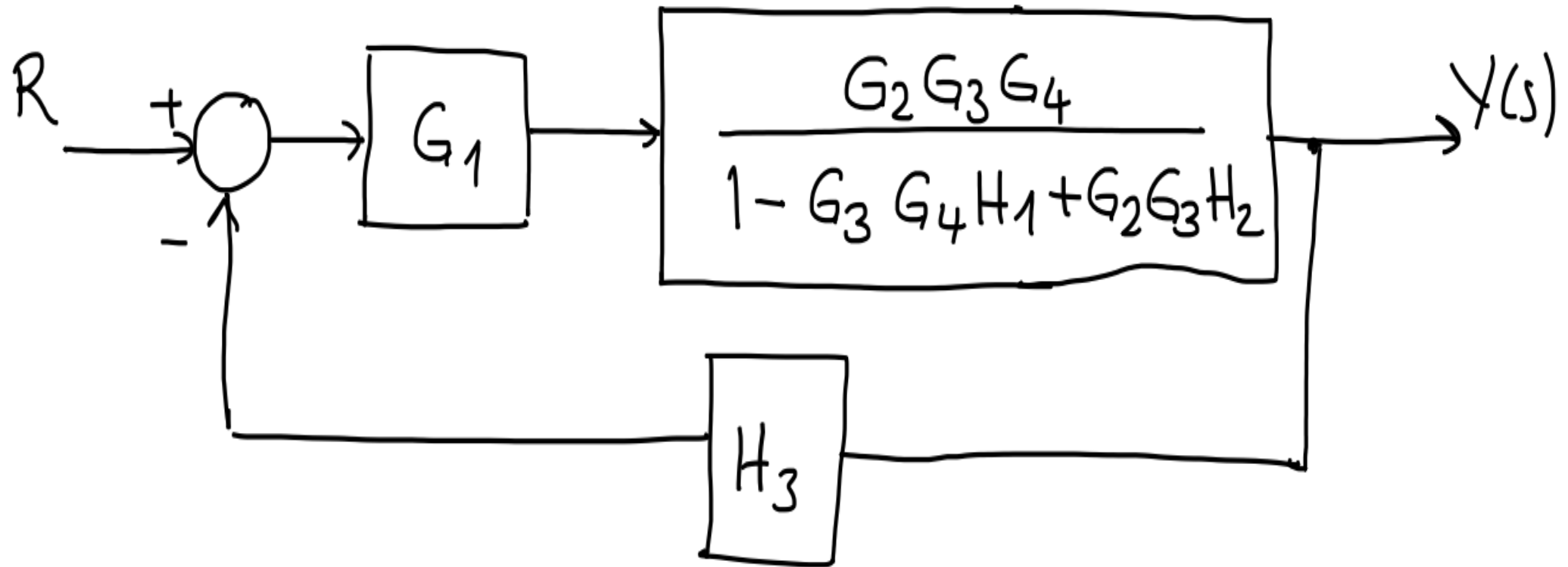
$$\begin{bmatrix} Y_1(s) \\ Y_2(s) \end{bmatrix} = \begin{bmatrix} G_{11}(s) & G_{12}(s) \\ G_{21}(s) & G_{22}(s) \end{bmatrix} \begin{bmatrix} R_1(s) \\ R_2(s) \end{bmatrix}$$

MIMO system



$\frac{Y(s)}{R(s)} = ?$ (use block diagram reduction)





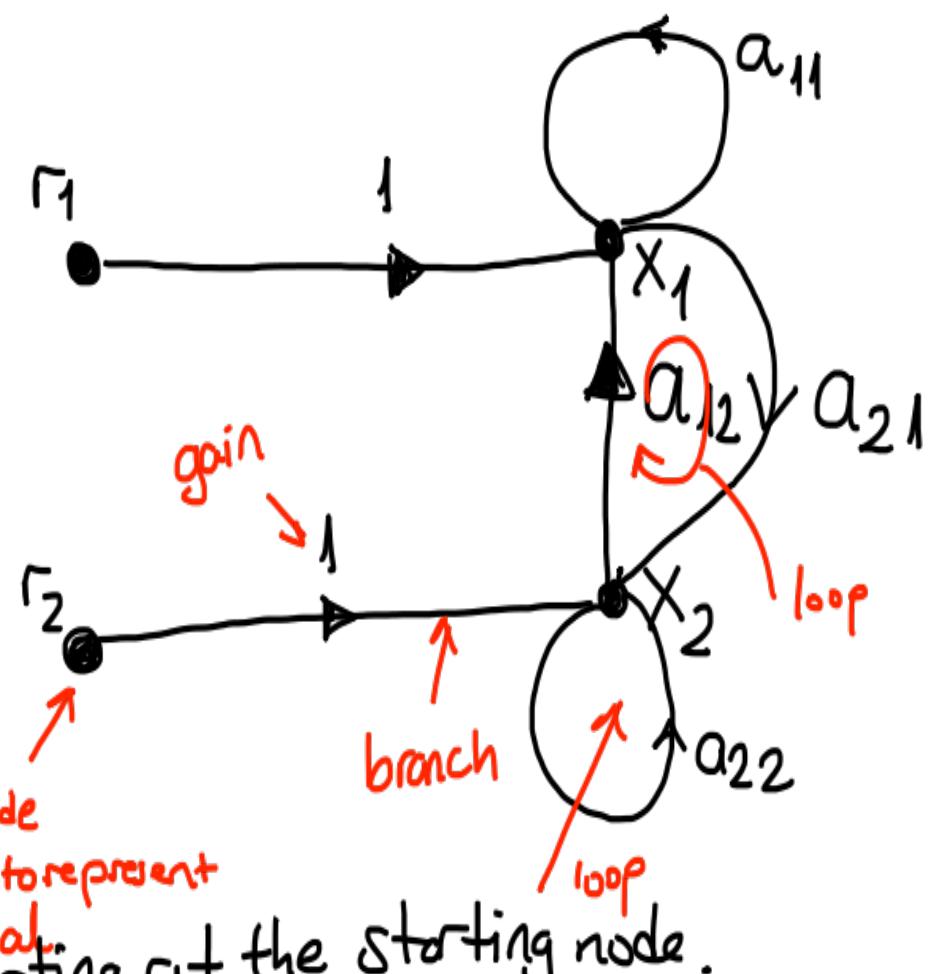
Signal Flow Graph Models

- For a system with reasonably complex interconnections, the block diagram reduction process is cumbersome.
- An alternative method for determining the relationship between system variables has been developed by Mason is based on graph theory
- A signal flow graph is a diagram consisting "nodes" that are connected by directed branches and it is a graphical representation of a system of linear relations.

EX:

$$X_1 = a_{11}X_1 + a_{12}X_2 + r_1$$

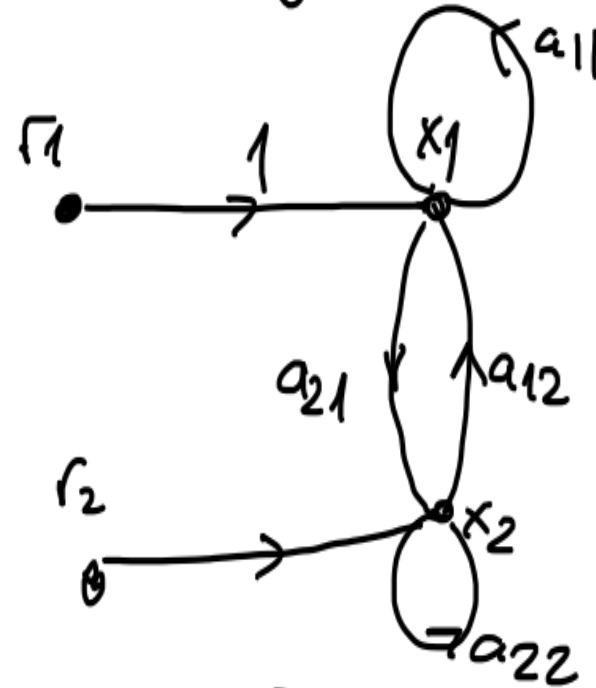
$$X_2 = a_{21}X_1 + a_{22}X_2 + r_2$$



- A "loop" is a closed path terminating at the starting node.
- Each signal or a variable is represented by a "NODE".
- Path between two nodes is called a "BRANCH".
- The value attached to each branch is called a "GAIN".
- A continuous sequence of branches that can be traversed from one signal to another is called a "PATH".

Δ_k : The Δ corresponding to that part of the graph that is non-touching with the k^{th} forward path.

EX: $X_1 = a_{11}x_1 + a_{12}x_2 + r_1$
 $X_2 = a_{21}x_1 + a_{22}x_2 + r_2$



$\frac{X_1}{r_1} = ?$

$\delta_1 = 1$

$M = \frac{X_1}{r_1} = \sum_{k=1}^N \frac{\delta_k \Delta_k}{\Delta} \rightarrow N=1 \Rightarrow M = \frac{X_1}{r_1} = \frac{\delta_1 \Delta_1}{\Delta}$

$\Delta = 1 - (a_{11} + a_{21} \cdot a_{12} + a_{22}) + (a_{11} \cdot a_{22})$

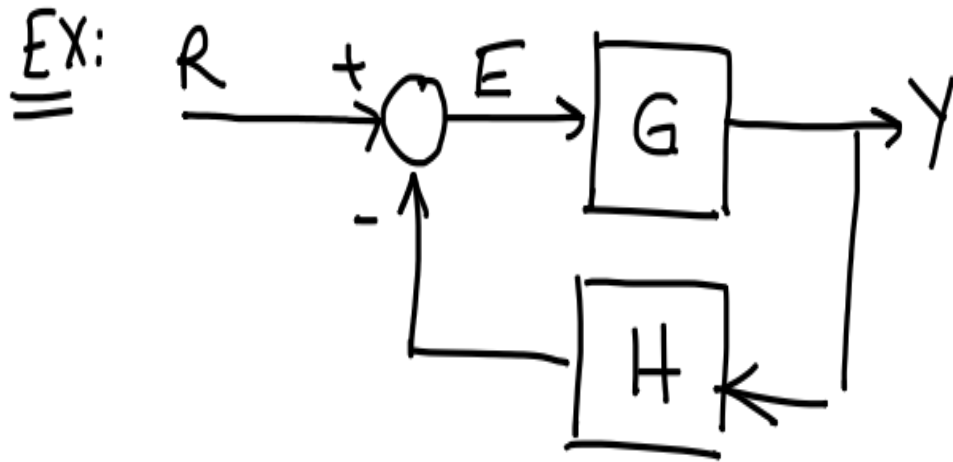
$\Delta_1 = 1 - (a_{22})$

$$M = \frac{X_1}{r_1} = \frac{\Delta_1 \delta_1}{\Delta} = \frac{(1-a_{22})1}{1 - (a_{11} + a_{12}a_{21} + a_{22}) + a_{11}a_{12}}$$

HW=? show that $\frac{X_1}{r_2} = \frac{a_{12}}{\Delta}$, $\frac{X_2}{r_1} = \frac{a_{21}}{\Delta}$

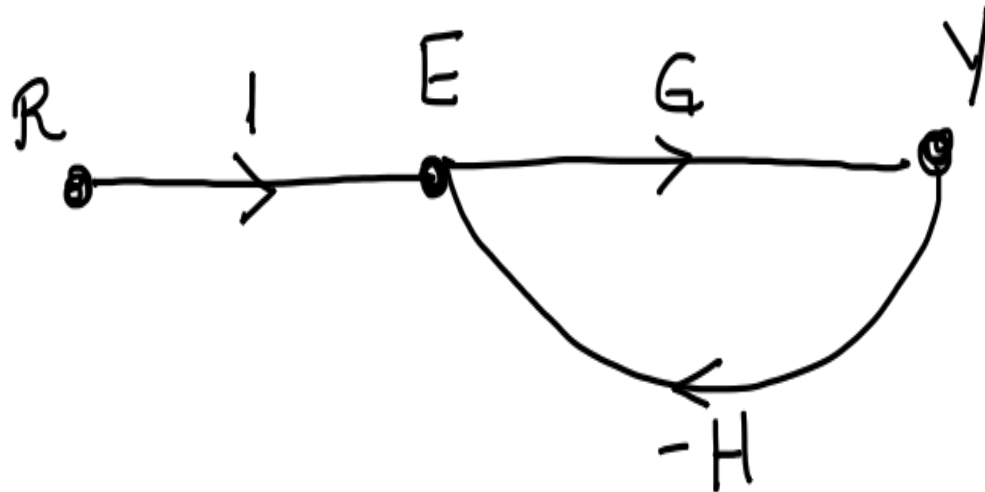
$$\frac{X_2}{r_2} = \frac{1-a_{11}}{\Delta}$$

$$\begin{bmatrix} X_1 \\ X_2 \end{bmatrix} = \frac{1}{\Delta} \begin{bmatrix} 1-a_{22} & a_{12} \\ a_{21} & 1-a_{11} \end{bmatrix} \begin{bmatrix} r_1 \\ r_2 \end{bmatrix}$$



$$\frac{Y}{R} = ?$$

$$\frac{Y}{R} = \frac{G}{1+GH}$$



$$\frac{Y}{R} = \sum_{k=1}^N \frac{\delta_k \Delta_k}{\Delta}$$

$$N=1$$

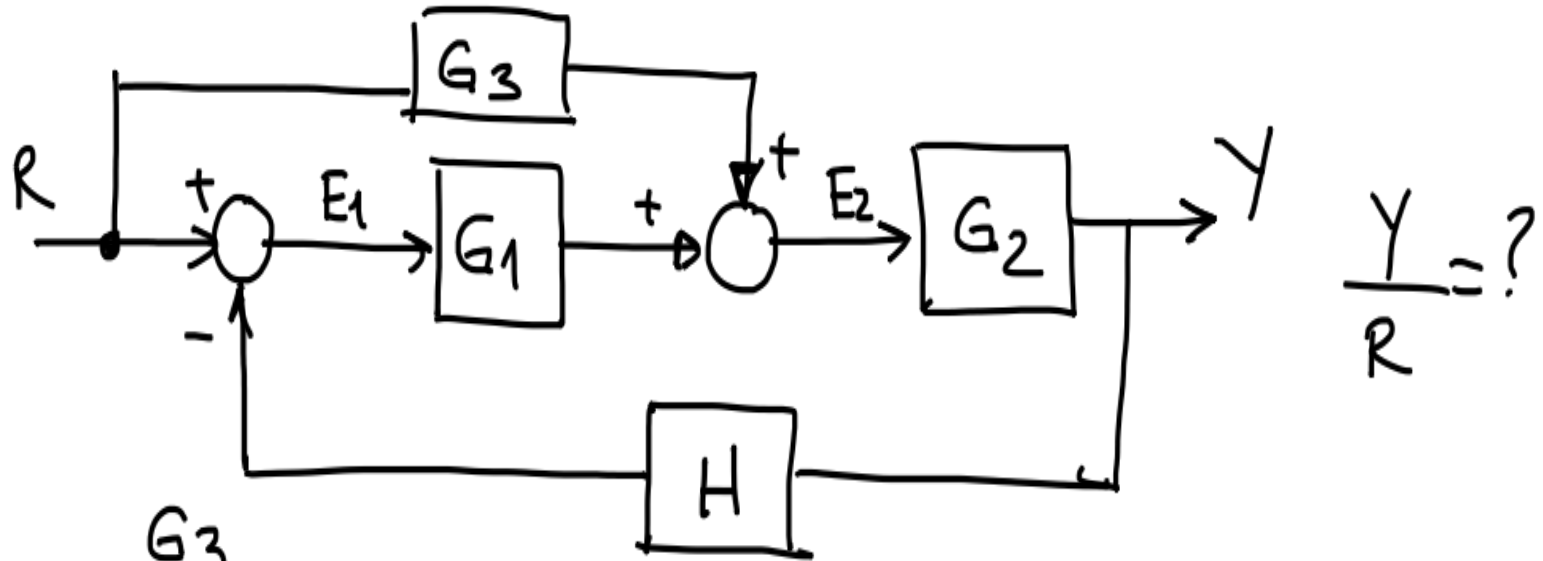
$$\delta_1 = G$$

$$\Delta = 1 - (-GH)$$

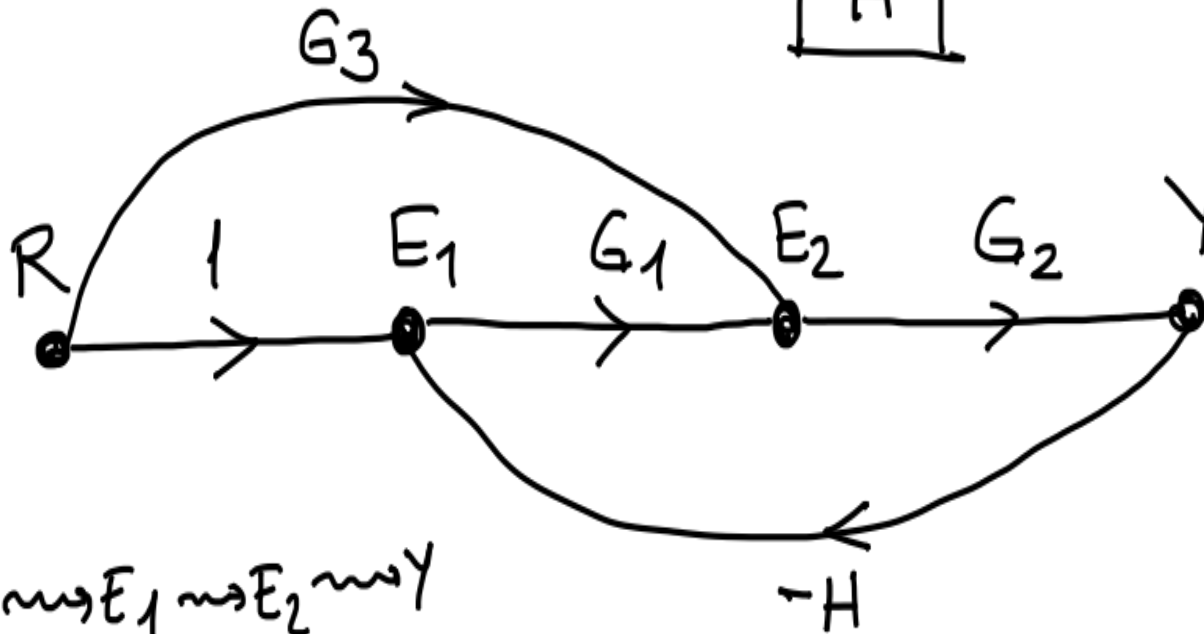
$$\Delta_1 = 1$$

$$\frac{Y}{R} = \frac{\delta_1 \Delta_1}{\Delta} = \frac{G \cdot 1}{1+GH} = \frac{G}{1+GH}$$

Ex:



$$\frac{Y}{R} = ?$$



$$\frac{Y}{R} = \sum_{k=1}^N \frac{\delta_k \Delta_k}{\Delta}$$

$$N=2$$

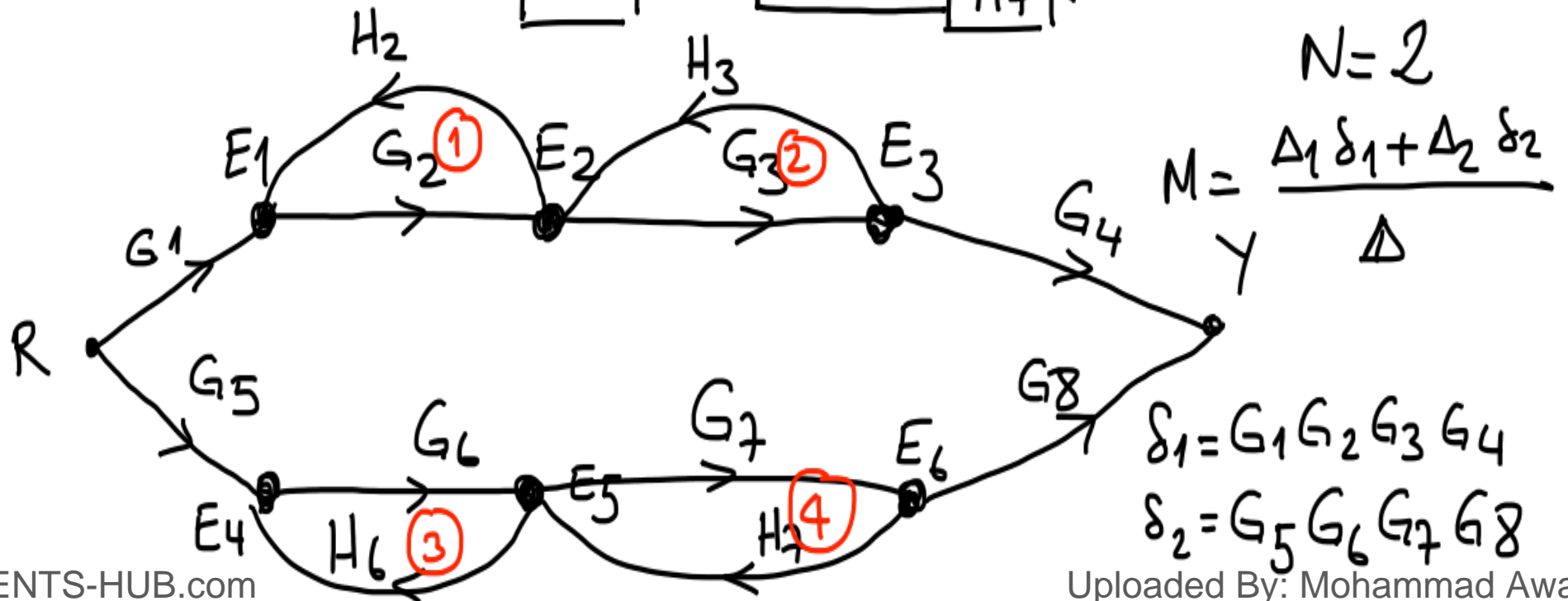
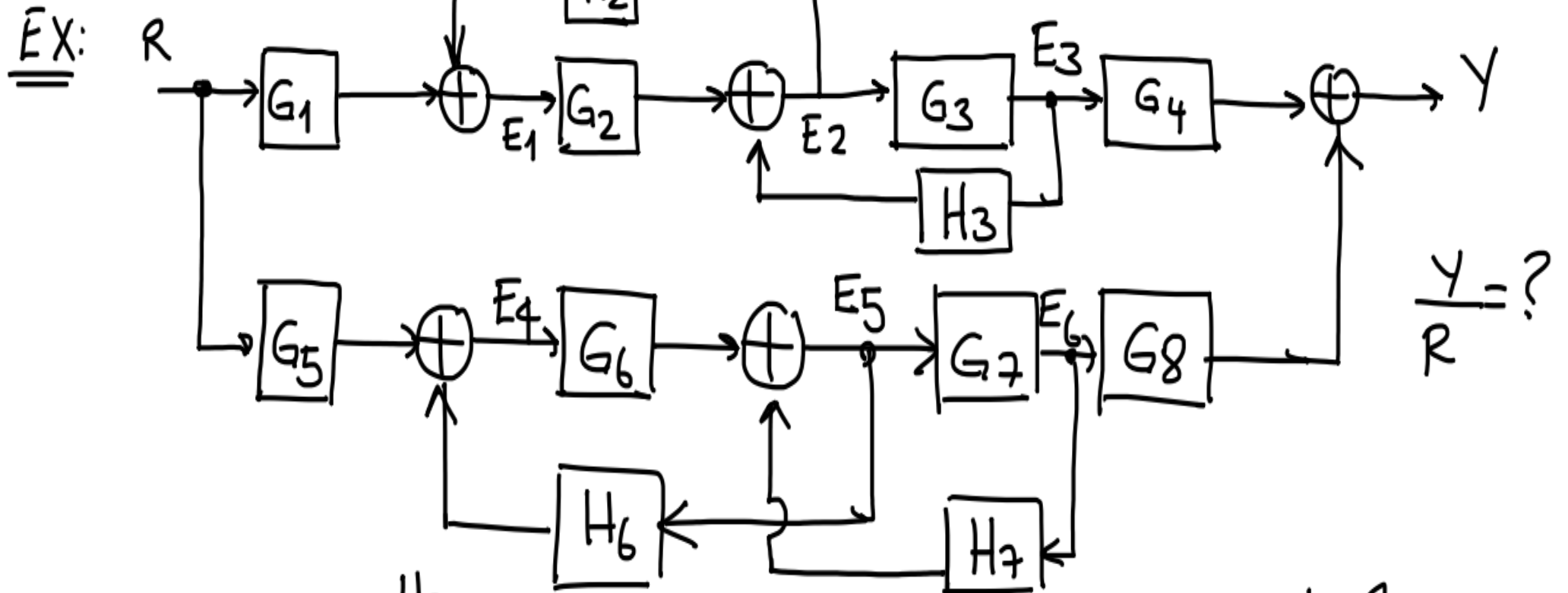
$$\frac{Y}{R} = \frac{\delta_1 \Delta_1 + \delta_2 \Delta_2}{\Delta}$$

① $R \rightsquigarrow E_1 \rightsquigarrow E_2 \rightsquigarrow Y$

② $R \rightsquigarrow E_2 \rightsquigarrow Y$

$$\delta_1 = G_1 G_2, \quad \delta_2 = G_3 G_2, \quad \Delta = 1 - (-G_1 G_2 H), \quad \Delta_1 = 1, \quad \Delta_2 = 1$$

$$\frac{Y}{R} = \frac{G_1 G_2 + G_3 G_2}{1 + G_1 G_2 H}$$



$$\Delta = 1 - (G_2 H_2 + G_3 H_3 + G_6 H_6 + G_7 H_7) \\ + (G_2 H_2 G_6 H_6 + G_2 H_2 G_7 H_7 + G_3 H_3 G_6 H_6 + G_3 H_3 G_7 H_7)$$

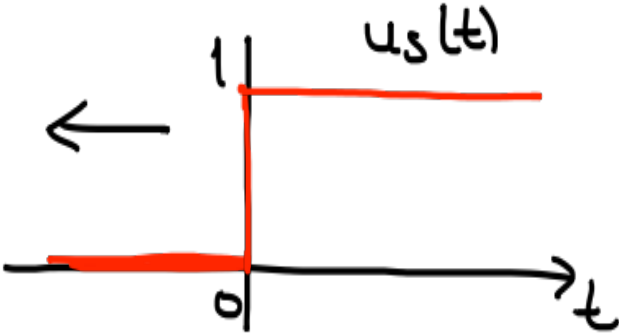
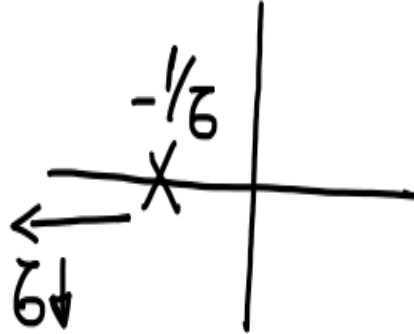
$$\Delta_1 = 1 - (G_6 H_6 + G_7 H_7)$$

$$\Delta_2 = 1 - (G_2 H_2 + G_3 H_3)$$

$$M = \frac{Y}{R} = \frac{\left[1 - (G_6 H_6 + G_7 H_7) \right] G_1 G_2 G_3 G_4 + (1 - G_2 H_2 - G_3 H_3) G_5 G_6 G_7 G_8}{\Delta}$$

First Order Dynamic Systems

Consider $\frac{Y(s)}{R(s)} = \frac{1}{\tau s + 1} = G(s)$ $\xrightarrow{R(s)}$ $\boxed{\frac{1}{\tau s + 1}}$ $\rightarrow Y(s)$

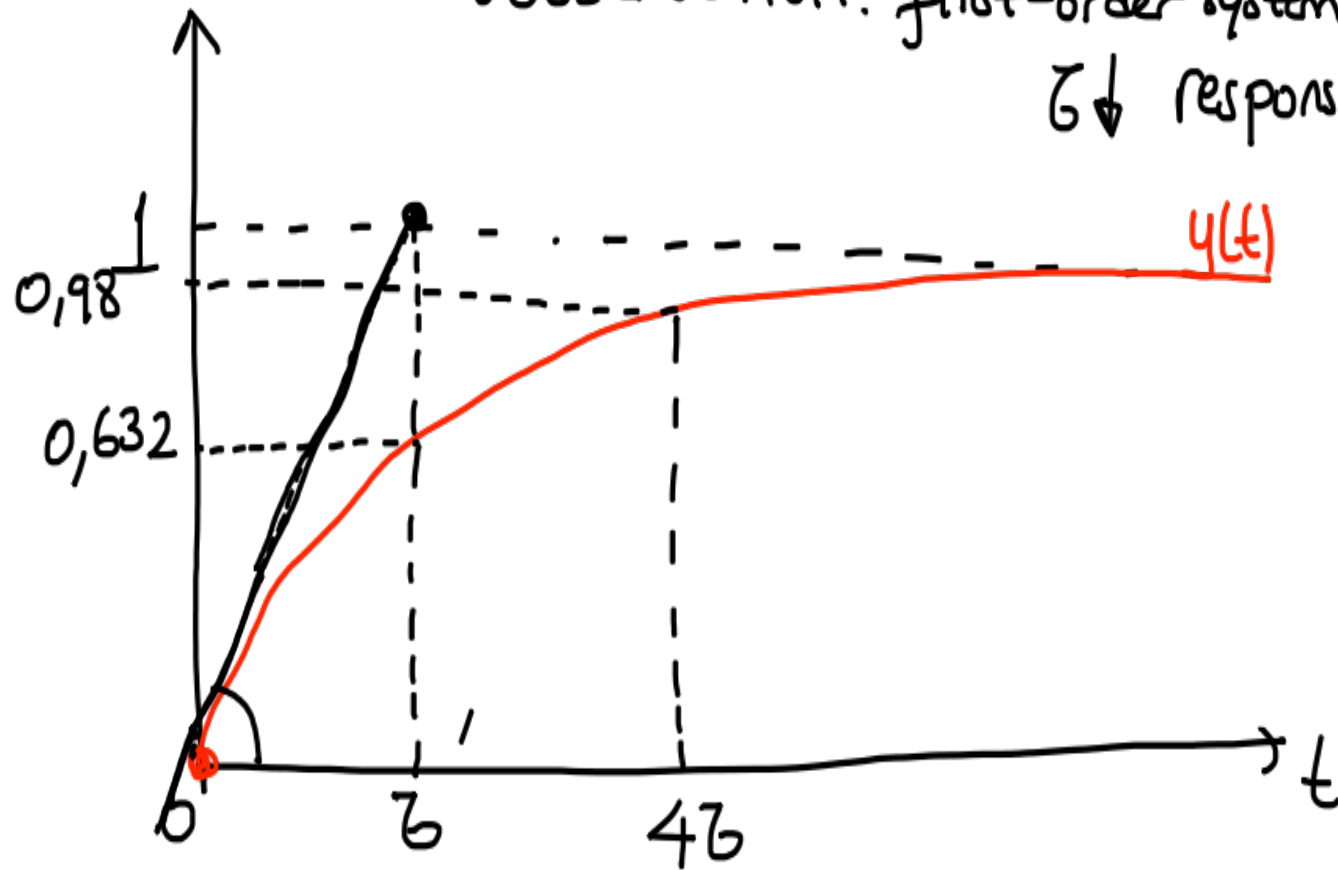
Assume $R(s) = \frac{1}{s}$ \leftarrow  

$$Y(s) = G(s)R(s) = \frac{1}{\tau s + 1} \cdot \frac{1}{s} = \frac{A=1}{s} + \frac{B=-\tau}{\tau s + 1}$$

$$= \frac{1}{s} - \frac{1}{s + 1/\tau}$$

$$y(t) = \mathcal{L}^{-1}\{Y(s)\} = (1 - e^{-1/\tau t})u_s(t)$$

- observation: first-order systems can't make overshoots.
 $\tau \downarrow$ response speed \uparrow



$$y(0) = 0$$

$$y(\infty) = 1$$

$$y(t) = (1 - e^{-\frac{1}{\tau}t}) u_s(t)$$

$$t = \tau \rightarrow y(\tau) = ?$$

$$y(\tau) = (1 - e^{-1}) = 0,632$$

$$y(4\tau) = (1 - e^{-4}) = 0,98$$

τ = time-constant

$t \geq 4\tau$ steady-state

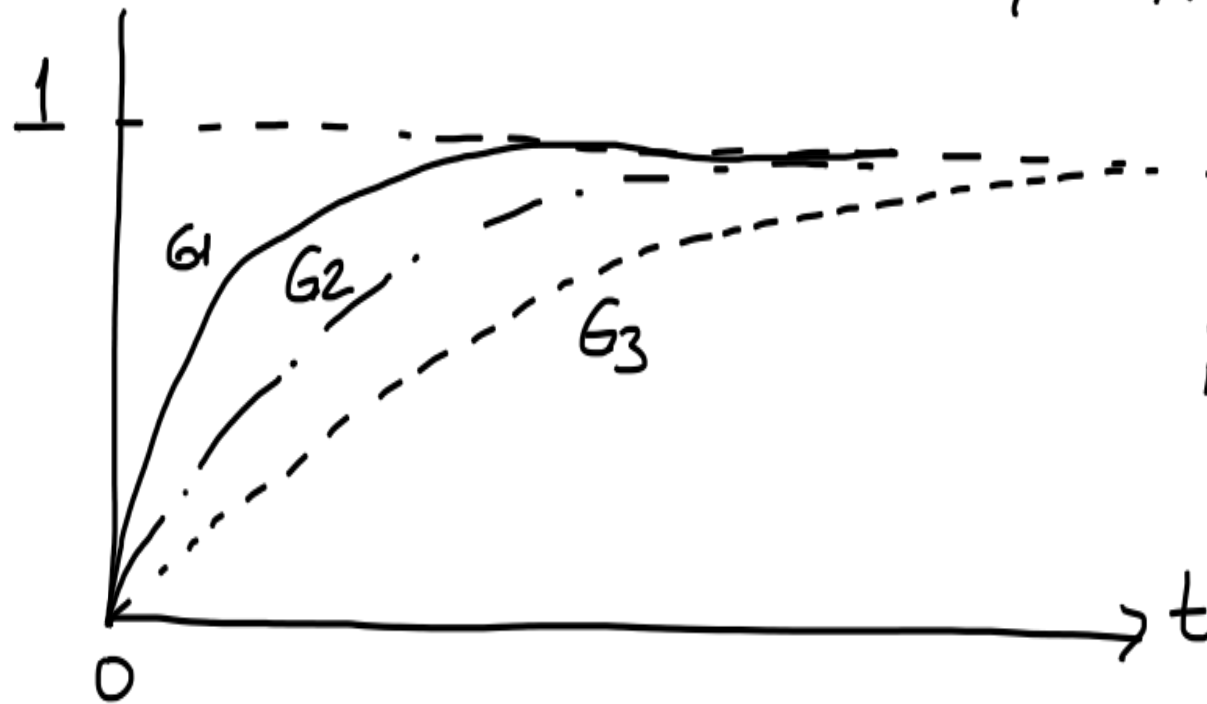
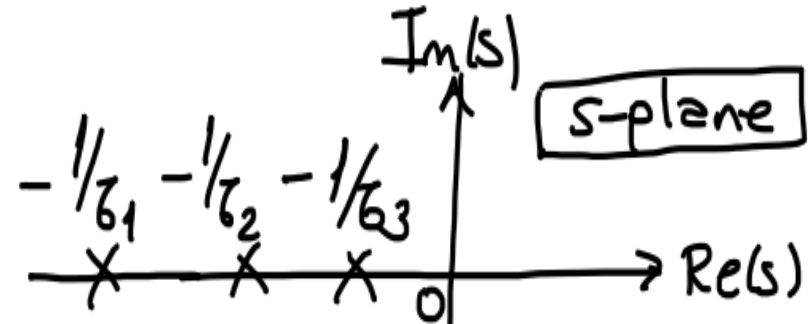
$$\left. \frac{d}{dt} y(t) \right|_{t=0} = \left. \frac{1}{\tau} e^{-\frac{1}{\tau}t} \right|_{t=0} = \frac{1}{\tau}$$

$$G_1 = \frac{1}{\tau_1 s + 1}$$

$$G_2 = \frac{1}{\tau_2 s + 1}$$

$$G_3 = \frac{1}{\tau_3 s + 1}$$

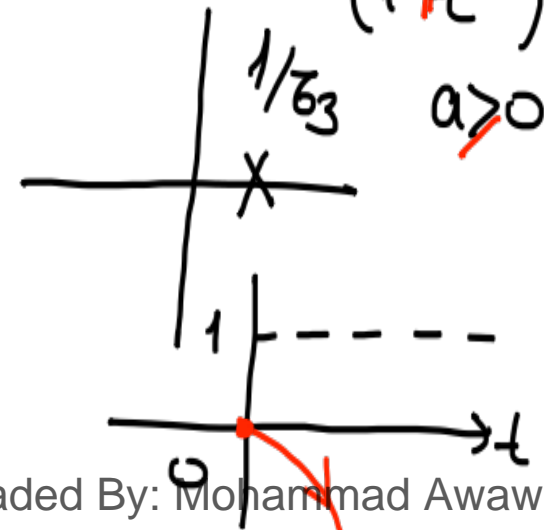
$$\tau_3 > \tau_2 > \tau_1$$



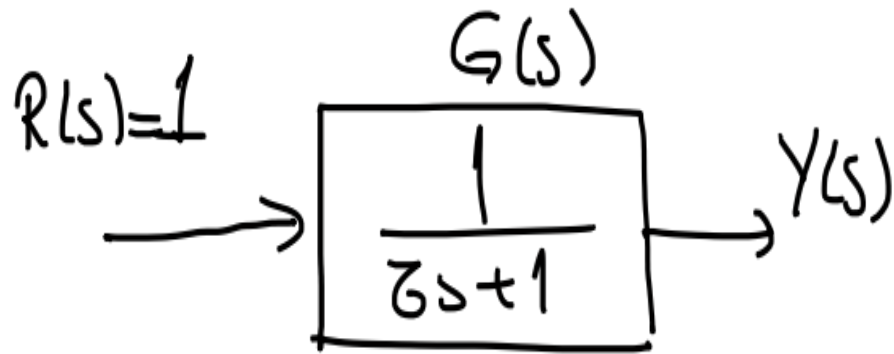
if $\tau_3 < 0$

$$y(t) = \left(1 - e^{-\frac{1}{\tau_3} t}\right) u(t)$$

$$= (1 + e^a) u(t) \quad a > 0$$



Unit Impulse Response of a 1st order System:



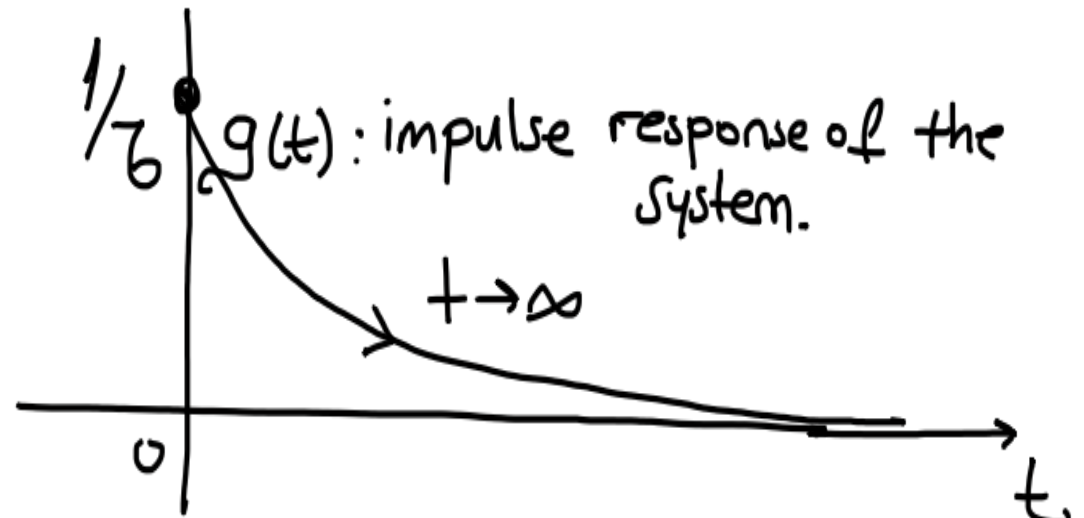
$$Y(s) = G(s)R(s) \quad R(s) = 1$$

$$Y(s) = G(s)$$

$$Y(s) = G(s) = \frac{1}{\tau s + 1} = \frac{1/\tau}{s + 1/\tau}$$

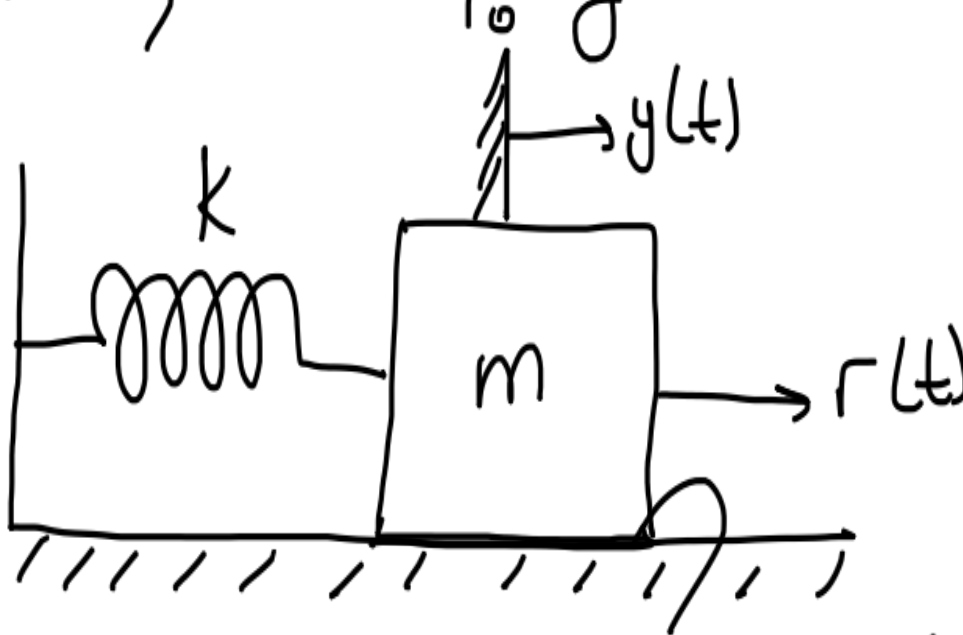
$$y(t) = \frac{1}{\tau} e^{-1/\tau t} \cdot u_s(t)$$

↓
 $g(t)$



2nd Order Systems

Consider a spring-mass system with mass m , friction b , and spring constant k



Eq'n of motion:

$$m\ddot{y} = r - ky - b\dot{y}$$

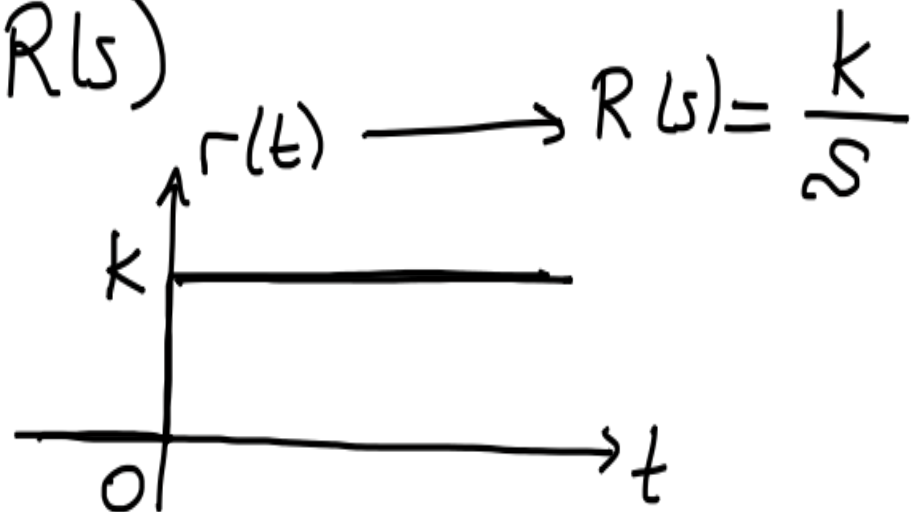
b : viscous friction const.

Let's take the Laplace Transform of both sides by assuming all initial cond'ns are identical to zero.
i.e., $y(0) = \dot{y}(0) = 0$

$$m\ddot{y} = r - ky - b\dot{y} \xrightarrow{\mathcal{L}} ms^2 Y(s) = R(s) - kY(s) - bsY(s)$$

$$\Rightarrow Y(s) = \frac{1}{ms^2 + bs + k} \cdot R(s)$$

Let's apply a constant input



$$Y(s) = \frac{1}{ms^2 + bs + k} \cdot \frac{k}{s} = \frac{k/m}{(s^2 + b/m s + k/m)s}$$

$$y(\infty) = \lim_{s \rightarrow 0} sY(s) = 1$$

$$\mathcal{L}^{-1} \rightarrow y(t) = ?$$

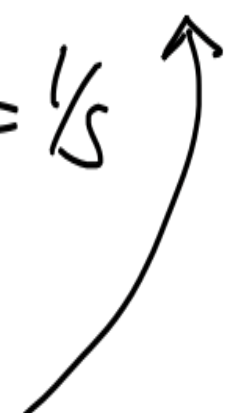
$$\frac{k}{m} =: \omega_n^2$$

$$\frac{b}{m} =: 2\zeta\omega_n$$

ζ : damping ratio
 ω_n : natural frequency.

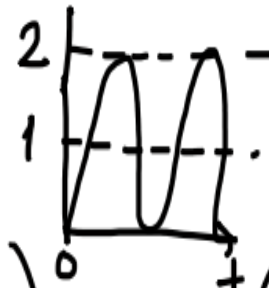
$$Y(s) = \frac{\omega_n^2}{s(s^2 + 2\zeta\omega_n s + \omega_n^2)}$$

$$\Rightarrow \frac{Y(s)}{R(s)} = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

if $R(s) = 1/s$ 

prototype second order system.

When ζ and ω_n varies, the response $y(t)$ changes...



$$y(t) = \begin{cases} 1 - \cos(\omega_n t) & (\xi = 0) \text{ (undamped case)} \\ 1 - e^{-\omega_n t} - \omega_n t \cdot e^{-\omega_n t} & \text{Graph of critically damped response} \quad (\xi = 1) \text{ (critically damped)} \\ 1 - \frac{\xi}{\alpha} e^{-\xi \omega_n t} \cosh(\omega_n \alpha t) & \text{Graph of overdamped response} \quad (\xi > 1) \text{ (overdamped resp)} \\ & \alpha = \sqrt{\xi^2 - 1} \\ 1 - \frac{1}{\beta} e^{-\xi \omega_n t} \sin(\omega_n \beta t + \phi) & \text{Graph of underdamped response} \quad (0 < \xi < 1) \text{ (underdamped case)} \\ & \beta = \sqrt{1 - \xi^2} \quad \phi = \cos^{-1} \xi \end{cases}$$

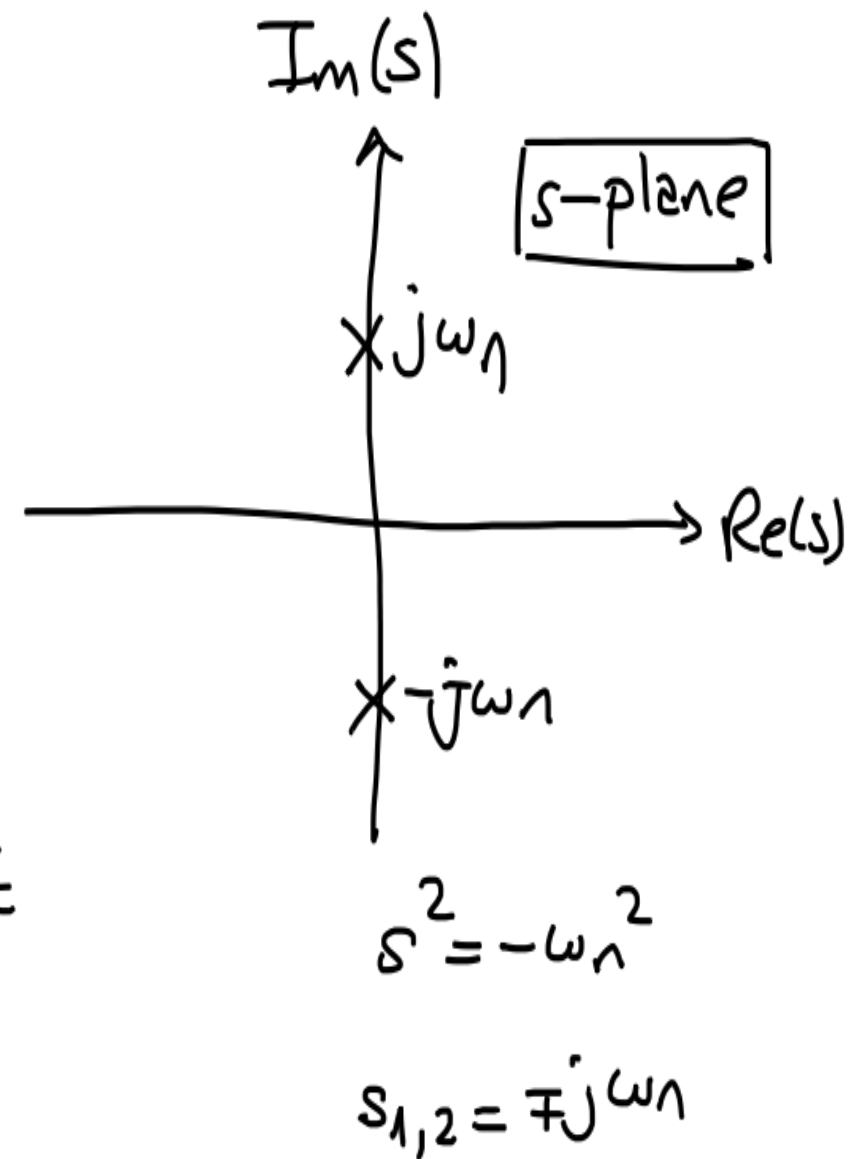
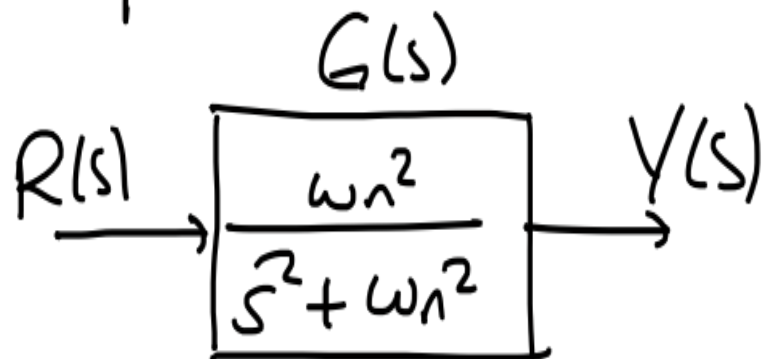
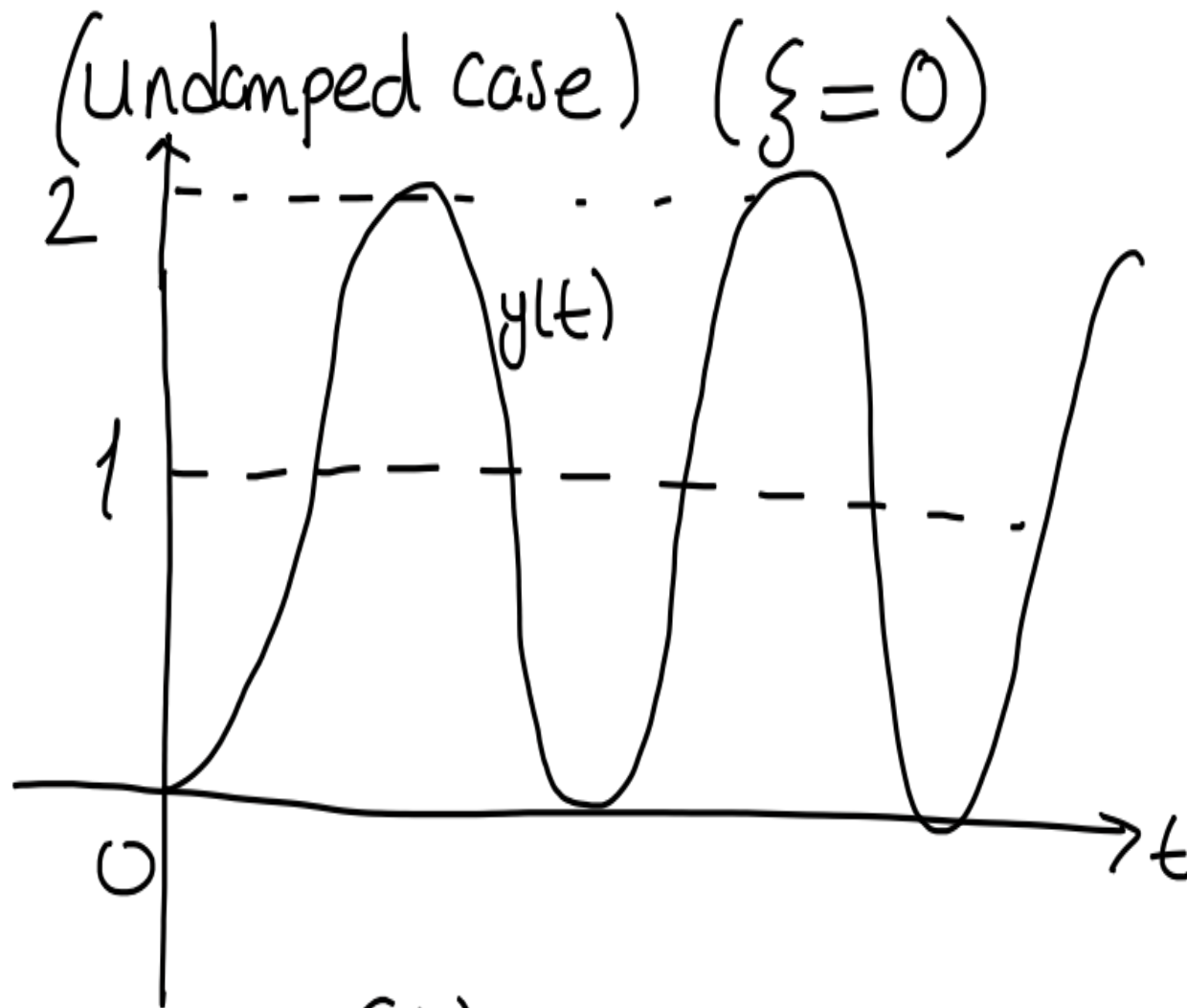
For underdamped case:

$$Y(s) = \frac{1}{s} - \frac{s + 2\xi\omega_n}{s^2 + 2\xi\omega_n s + \omega_n^2} = \frac{1}{s} - \frac{s + \xi\omega_n}{(s + \xi\omega_n)^2 + \omega_n^2 \sqrt{1 - \xi^2}} -$$

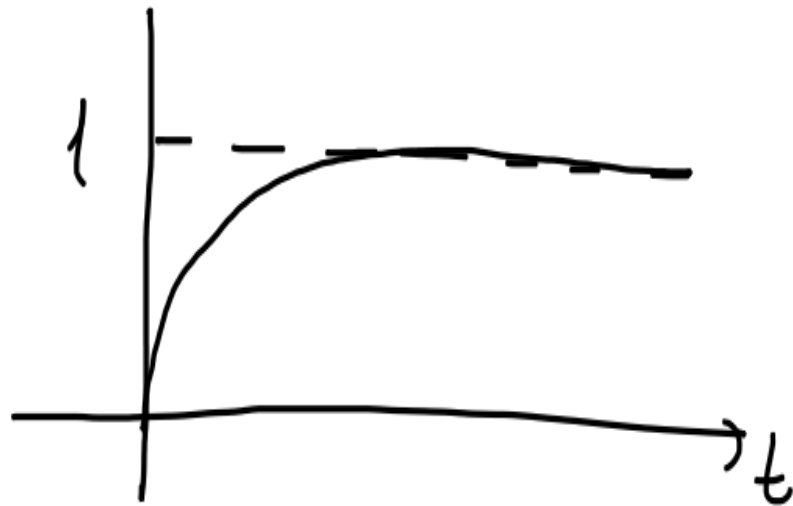
$$- \frac{\xi\omega_n}{(s + \xi\omega_n)^2 + \omega_n^2 \sqrt{1 - \xi^2}}$$

$$y(t) = 1 - e^{-\xi\omega_n t} \cdot \cos \omega_n \sqrt{1 - \xi^2} \cdot t$$

$$- e^{-\xi\omega_n t} \cdot \frac{\xi}{\sqrt{1 - \xi^2}} \cdot \sin \omega_n \sqrt{1 - \xi^2} t$$

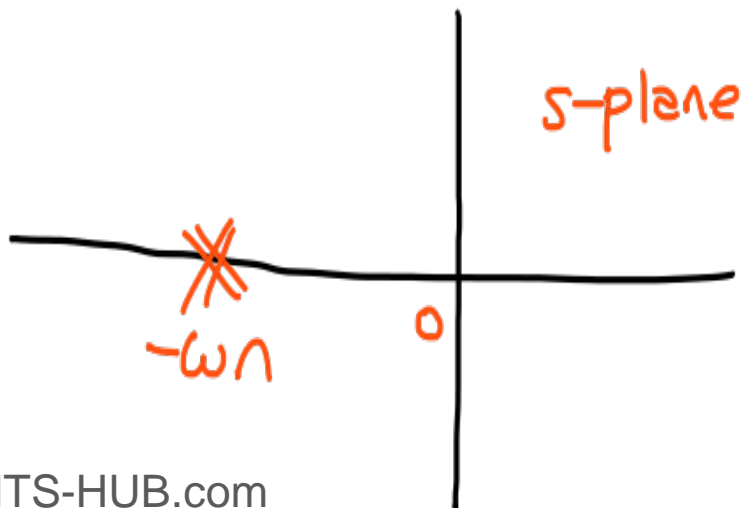


(Critically damped case, $\zeta=1$)



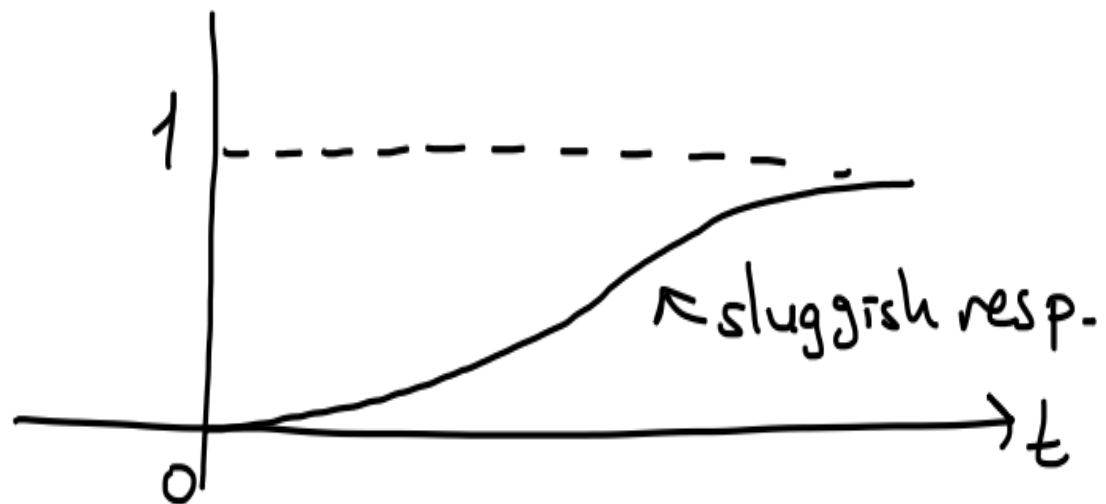
← Fastest resp. w/o overshoot.
(very similar to the resp. of a first order syst)

$$G(s) = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$



$$G(s) = \frac{\omega_n^2}{s^2 + 2\omega_n s + \omega_n^2} \quad \zeta=1 = \frac{\omega_n^2}{(s + \omega_n)^2}$$

(overdamped case) ($\zeta > 1$)

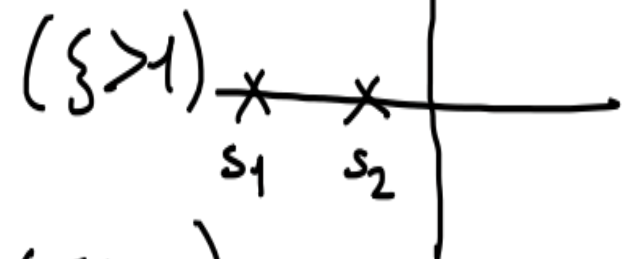


$$G(s) = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

$$\Delta(s) = s^2 + 2\zeta\omega_n s + \omega_n^2 = 0$$

↑
characteristic poly.

$$s_{1,2} = -\zeta\omega_n \mp \sqrt{\zeta^2\omega_n^2 - \omega_n^2}$$



$$s_{1,2} = -\zeta\omega_n \mp \omega_n \sqrt{\zeta^2 - 1}$$

($\zeta > 1$)

$$s_1 = -\zeta\omega_n - \omega_n \sqrt{\zeta^2 - 1}$$

$$s_2 = -\zeta\omega_n + \omega_n \sqrt{\zeta^2 - 1}$$